

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

A25SO
Revision 7
Piper
PA-46-310P
PA-46-350P

March 4, 1998

TYPE CERTIFICATE DATA SHEET NO. A25SO

This data sheet, which is part of Type Certificate No. A25SO, prescribes conditions and limitations under which the product for which type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder: The New Piper Aircraft, Inc.
 2926 Piper Drive
 Vero Beach, Florida 32960

I. - Model PA-46-310P (Malibu), 6 PCLM (Normal Category), Approved September 27, 1983.

Engine Teledyne Continental Model TSI0-520-BE

Fuel 100/100LL minimum grade aviation gasoline

Engine Limits For all operations:
 2600 RPM and 38" Hg MAP (310 HP), sea level to 24,000 ft.
 2600 RPM and 35" Hg MAP above 24,000 ft.

 2400 RPM and 31" Hg MAP maximum when leaned to 50° F lean of peak,
 any altitude.

Propeller and Propeller Limits Hartzell, Hub BHC-C2YF-1BF, Blade F8052 ()
Pitch: High 38.0° ± 1°, Low 16.0° ± 0.2 at 30" station.
Diameter: Not over 80", not under 78".
Spinner: Hartzell D-4810 or D-4810P.
Governor: Hartzell Model E-5-2.

Airspeed Limits

V _{NE} (Never Exceed)	203 KIAS
V _{NO} (Maximum Structural Cruise)	173 KIAS
V _A (Maneuvering 4100 lb.)	135 KIAS
V _A (Maneuvering 2450 lb.)	103 KIAS
V _{FE} (Maximum Flaps Extended)	120 KIAS
V _{LO} (Maximum Landing Gear Operation)	
Extension	170 KIAS
Retraction	130 KIAS
V _{LE} (Maximum Landing Gear Extended)	200 KIAS

<u>C.G. Range (Gear Extended)</u>	<u>WT.</u>	<u>FWD. LIMIT</u>	<u>AFT LIMIT</u>
	<u>(LB.)</u>	<u>IN. AFT OF DATUM</u>	<u>IN. AFT OF DATUM</u>
	4100	143.3 in.	147.1 in.
	3680	136.1 in.	147.1 in.
	2450 or less	130.7 in.	147.1 in.

Empty Weight C.G. Range None

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<u>Maximum Weight</u>	Ramp - 4118 lb. Takeoff - 4100 lb. Landing - 3900 lb.																																				
<u>No. of Seats</u>	6 (2 at +135.5, 2 at +177.0, 2 at +218.75)																																				
<u>Maximum Baggage</u>	100 lb. at (+88.6) (Fwd.) 100 lb. at (+248.23) (Aft)																																				
<u>Fuel Capacity</u>	122 gals. at (+150.31) (2 wing tanks) See NOTE 1 for data on system fuel.																																				
<u>Oil Capacity</u>	8 qts. at (+53.5) See NOTE 1 for data on system oil.																																				
<u>Maximum Operating Altitude</u>	25,000 ft.																																				
<u>Maximum Cabin Operating Pressure Differential</u>	5.5 PSID																																				
<u>Control Surface Movements</u>	<table> <tr> <td>Ailerons</td> <td>($\pm 1^\circ$)</td> <td>Up</td> <td>18°</td> <td>Down</td> <td>18°</td> </tr> <tr> <td>Elevator</td> <td>($\pm 5^\circ$)</td> <td>Up</td> <td>23.5°</td> <td>Down</td> <td>14.5°</td> </tr> <tr> <td>Rudder</td> <td>($\pm 1^\circ$, -0°)</td> <td>Left</td> <td>26°</td> <td>Right</td> <td>30°</td> </tr> <tr> <td>Elevator</td> <td>($\pm 0^\circ$, -1°)</td> <td>Down</td> <td>24.5°</td> <td>Up</td> <td>19°</td> </tr> <tr> <td>Trim Tab</td> <td></td> <td></td> <td></td> <td>(Elevator Neutral)</td> <td></td> </tr> <tr> <td>Wing Flaps</td> <td>($\pm 1^\circ$)</td> <td>Up</td> <td>0°</td> <td>Down</td> <td>35°</td> </tr> </table>	Ailerons	($\pm 1^\circ$)	Up	18°	Down	18°	Elevator	($\pm 5^\circ$)	Up	23.5°	Down	14.5°	Rudder	($\pm 1^\circ$, -0°)	Left	26°	Right	30°	Elevator	($\pm 0^\circ$, -1°)	Down	24.5°	Up	19°	Trim Tab				(Elevator Neutral)		Wing Flaps	($\pm 1^\circ$)	Up	0°	Down	35°
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Trim Tab				(Elevator Neutral)																																	
Wing Flaps	($\pm 1^\circ$)	Up	0°	Down	35°																																
<u>Manufacturer's Serial Numbers</u>	46-8408001 through 46-8408087, 46-8508001 through 46-8508109, 46-8608001 through 46-8608067, 4608001 through 4608140.																																				

II. - Model PA-46-350P (Malibu Mirage), 6 PCLM (Normal Category), Approved August 30, 1988.

<u>Engine</u>	Textron Lycoming Model TI0-540-AE2A
<u>Fuel</u>	100/100LL minimum grade aviation gasoline
<u>Engine Limits</u>	For all operations: 2500 RPM and 42.0" Hg MAP (350 HP), sea level to 20,600 ft. 42 - 1.6" Hg MAP decrease per each 1000 ft. altitude increase, 20,600 ft. to 25,000 ft.
<u>Propeller and Propeller Limits</u>	<p>Hartzell, Hub HC-I2YR-1BF, Blade F8074 (standard 2 blade) Pitch: High $40.5^\circ \pm 0.5^\circ$ Low $17.6^\circ \pm 0.2^\circ$ at 30" station. Diameter: Not over 80", not under 79". Spinner: Hartzell A-2298-3P. Governor: Hartzell Model V-5-2.</p> <p>Hartzell, Hub HC-13YR-1E, Blade 7890K (optional 3 blade) Pitch: High $38.7^\circ \pm 0.5^\circ$ Low $13.65^\circ \pm 0.15^\circ$ at 30" station. Diameter: 80" Spinner: Hartzell D-6750. Governor: Hartzell Model V-5-2.</p> <p>Do not exceed 36" MAP below 2400 RPM Do not exceed 32" MAP below 2300 RPM</p>

<u>Airspeed Limits</u>	V _{NE} (Never Exceed)		198 KIAS
	V _{NO} (Maximum Structural Cruise)		168 KIAS
	V _A (Maneuvering 4300 lb.)		133 KIAS
	V _A (Maneuvering 2450 lb.)		100 KIAS
	V _{FE} (Maximum Flaps Extended)		116 KIAS
	V _{LO} (Maximum Landing Gear Operation)		
	Extension		165 KIAS
Retraction		126 KIAS	
V _{LE} (Maximum Landing Gear Extended)		195 KIAS	
<u>C.G. Range (Gear Extended)</u>	WT. (LB.)	FWD. LIMIT IN. AFT OF DATUM	AFT LIMIT IN. AFT OF DATUM
	4300	143.3 in.	147.1 in.
	4100	139.1 in.	147.1 in.
	4000	137.0 in.	146.5 in.
	2450	130.7 in.	137.6 in.
	2400	130.7 in.	137.3 in.
<u>Empty Weight C.G. Range</u>	None		
<u>Maximum Weight</u>	Ramp - 4318 lb. Takeoff - 4300 lb. Landing - 4100 lb.		
<u>No. of Seats</u>	6 (2 at +135.5, 2 at +177.0, 2 at +218.75)		
<u>Maximum Baggage</u>	100 lb. at (+88.6) (Fwd.) 100 lb. at (+248.23) (Aft)		
<u>Fuel Capacity</u>	122 gals. at (+150.31) (2 wing tanks) See NOTE 1 for data on system fuel.		
<u>Oil Capacity</u>	12 qts. at (+53.5) See NOTE 1 for data on system oil.		
<u>Maximum Operating Altitude</u>	25,000 ft.		
<u>Maximum Cabin Operating Pressure Differential</u>	5.5 PSID		
<u>Control Surface Movements</u>	Ailerons	(±1°)	Up 18° Down 18°
	Elevator	(±0.5°)	Up 23.5° Down 14.5°
	Rudder	(±1°, -0°)	Left 26° Right 30°
	Elevator	(±0°, -1°)	Down 24.5° Up 19°
	Trim Tab		(Elevator Neutral)
	Wing Flaps	(±1°)	Up 0° Down 35°
<u>Manufacturer's Serial Numbers</u>	4622001 through 4622200, and 4636001 and up.		

DATA PERTINENT TO ALL MODELS

Datum	100 in. forward pressure bulkhead.
Leveling Means	Top or bottom fuselage at B.L. 0 (constant section).
Certification Basis	<p>Type Certificate No. A25SO issued September 27, 1983. Date of application for Type Certificate, August 22, 1979.</p> <p>FAR Part 23, effective February 1, 1965, as amended by Amendment 23-25, effective March 6, 1980; FAR 25.783(e) as amended by Amendment 25-54, effective October 14, 1980; FAR 25.831(c) and (d) as amended by Amendment 25-41, effective September 1, 1977; and FAR 36, Appendix F through Amendment 36-15, effective May 6, 1988 when equipped with 2 blade propeller or FAR 36, Appendix G through Amendment 36-16, effective December 18, 1988 when equipped with optional 3 blade propeller. No equivalent safety findings. Special Conditions No. 23-ACE-53, Docket No. 082CE.</p>
Production Basis	Production Certificate No. 206. Production Limitation Record issued and the manufacturer authorized to issue airworthiness certificate under the delegation option provisions of FAR 21.
Equipment	<p>The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the aircraft for certification.</p> <p>In addition, one of the following items of equipment are required:</p> <ol style="list-style-type: none"> 1. DOA No. SO-1 approved Airplane Flight Manual Piper Report FT 157, Appendix D or Pilot's Operating Handbook and FAA approved Airplane Flight Manual Report No. VB-1200 for Model PA-46-310P, S/N 46-8408001 through 46-8608067, and 4608001 through 4608007. 2. DOA No. SO-1 approved Pilot's Operating Handbook and FAA approved Airplane Flight Manual Report No. VB-1300 for Model PA-46-310P, S/N 4608008 through 4608140. 3. DOA No. SO-1 approved Pilot's Operating Handbook and FAA approved Airplane Flight Manual Report No. VB-1332 for Model PA-46-350P, S/N 4622001 through 4622200. 4. DOA No. SO-1 approved Pilot's Operating Handbook and FAA approved Airplane Flight Manual Report No. VB-1609 for Model PA-46-350P, S/N 4636001 through 4636020. 5. DOA No. SO-1 approved Pilot's Operating Handbook and FAA approved Airplane Flight Manual Report No. VB-1602 for Model PA-46-350P, S/N 4636021 through 4636131. 6. DOA No. SO-1 approved Pilot's Operating Handbook and FAA approved Airplane Flight Manual Report No. VB-1446 for Model PA-46-350P, S/N 4636132 and up.

Noise Characteristics

The corrected noise level of the Model PA-46-310P is 74.8 dBA at the Maximum Normal Operating Power at 2600 rpm. The noise level stated above has been approved by the Federal Aviation Administration in noise level test flights conducted in accordance with FAR 36, "Noise Standards: Aircraft Type and Airworthiness Certification." The aircraft noise is in compliance with FAR 36 noise standards applicable to this type.

The corrected noise level of the Model PA-46-350P is 74.7 dBA at the Maximum Normal Operating Power at 2500 rpm. The noise level stated above has been approved by the Federal Aviation Administration in noise level test flights conducted in accordance with FAR 36, "Noise Standards: Aircraft Type and Airworthiness Certification." The aircraft noise is in compliance with FAR 36 noise standards applicable to this type.

The corrected noise level of the Model PA-46-350P equipped with the optional 3 blade propeller is 79.7 dBA at the Maximum Normal Operating Power at 2500 rpm. The noise level stated above has been approved by the Federal Aviation Administration in noise level flight tests conducted in accordance with FAR 36, "Noise Standards: Aircraft Type and Airworthiness Certification." The aircraft noise is in compliance with FAR 36 noise standards applicable to this type.

NOTE 1. Current Weight and Balance Report, including list of equipment included in certificated empty weight and loading instructions when necessary, must be provided for each aircraft at the time of original certification.

The certified empty weight and corresponding center of gravity locations must include undrainable system oil (not included in oil capacity) and unusable fuel as noted below:

- (a) PA-46-310P
 - Fuel: 12 lb. at (+152.37)
 - Oil: 2.8 lb. at (+53.5)
- (b) PA-46-350P
 - Fuel: 12 lb. at (+152.37)
 - Oil: 3.8 lb. at (+61.0)

NOTE 2 All placards required in the POH and AFM must be installed in the appropriate locations. The following placard must be displayed in clear view of the pilot:

"The markings and placards installed in this airplane contain operating limitations which must be complied with when operating this airplane in the Normal Category. Other operating limitations which must be complied with when operating this airplane in this category are contained in the Airplane Flight Manual. No aerobatics maneuvers, including spins, approved."

NOTE 3

- (a) PA-46-310P
 - The life limit of the fuselage assembly, P/N 82250, is 10,145 hours time-in-service.
 - The life limit of the wing assembly, P/N 83100, is 15,580 hours time-in-service.
- (b) PA-46-350P
 - The life limit of the fuselage assembly, P/N 89600, is 10,145 hours time-in-service.
 - The life limit of the wing assembly, P/N 89640, is 15,580 hours time-in-service.

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