

DEPARTMENT OF TRANSPORTATION  
FEDERAL AVIATION ADMINISTRATION

A4SW Rev. No. 29 THRUSH AIRCRAFT, INC. (Snow, Rockwell, Ayres)		
600 S2D	S2R-R1340	S2R-G10
S2R	S2R-R1820	S2R-G5
S2R-T34	S2R-T65	S2R-G1
S2R-T15	S2RHG-T65	S2RHG-T34
S2R-R3S	S2R-T45	S2R-T660
S2R-T11	S2R-G6	
April 26, 2011		

**TYPE CERTIFICATE DATA SHEET NO. A4SW**

This data sheet which is a part of Type Certificate No. A4SW, prescribes conditions and limitations under which the product, for which the type certificate was issued, meets the airworthiness requirements of the Civil Air Regulations (CAR) or later Code of Federal Regulations (CFR).

Type Certificate Holder                      Thrush Aircraft, Inc.  
300 Old Pretoria Road  
P.O. Box 3149  
Albany, Georgia 31706-3149

Type Certificate Holder Record            Snow Aeronautical Company transferred TC to North American Rockwell Corporation on February 18, 1970  
North American Rockwell Corporation transferred TC to Rockwell International, Albany Aircraft Division on April 3, 1973  
Rockwell International, Albany Aircraft Division transferred TC to Rockwell International, Commander Aircraft Division on July 27, 1973  
Rockwell International, Commander Aircraft Division transferred TC to Ayres Corporation on November 28, 1977  
Ayres Corporation transferred TC to Quality Aerospace on November 26, 2001  
Quality Aerospace transferred TC to Thrush Aircraft, Inc. on July 9, 2003

**L-Model 600 S2D 1 PCLM (Restricted Category Only), Approved November 1, 1965**

Engine    Pratt & Whitney WASP R-1340-AN-1 (S3H1 Commercial designation) with carburetor parts list settings 395118-3 or A-18639-7

Fuel    80/87 minimum grade aviation gasoline

Engine Limits

	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>
Takeoff	600	2,250	36.0	S.L.
Max. Continuous	550	2,200	34.0	S.L.
Max. Continuous	550	2,200	32.5	5,000

Page No.	01	02	03	04	05	06	07	08	09	10	11	12	13	14	15	16	17
Rev. No.	29	29	29	29	29	29	29	29	29	29	29	29	29	29	29	29	29

Page No.	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34
Rev. No.	29	29	29	29	29	29	29	29	29	29	29	29	29	29	29	28	28

Page No.	35	36	37	38	39	40	41	42
Rev. No.	28	29	29	29	28	29	29	29

Propeller and Propeller Limits	Hamilton Standard, constant speed, 12D40 Hub, 6101-12 blades. Diameter 109 inches maximum, 107 inches minimum. Pitch settings, 11.5 <sup>o</sup> low and 27.0 <sup>o</sup> high at 42 inch station. Alternate settings, 11.5 <sup>o</sup> low and 21.5 <sup>o</sup> high at 42 inch station. Alternate blades, EAC AG100-2, settings 11.5 <sup>o</sup> low and 18 <sup>o</sup> high at 42 inches.		
Airspeed Limits (CAS) (See Note 2(n) for exceptions)	Vne (Never Exceed)	159 m.p.h. (138 knots)	
	Vp (Maneuvering)	126 m.p.h. (109 knots)	
	Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)	
C.G. Range	(+22.5) to (+29.0)		
Maximum Weight	6,000 lbs.		
Number of Seats	1 (+89.0)		
Maximum Cargo Load	See weight and balance data		
Fuel Capacity	109 gallons (+38.5) (100 gallon usable capacity, one 54.5 gallon tank in each wing, tanks interconnected). See NOTE 1 for data on unusable fuel.		
Oil Capacity	11.4 gallons total. 84 lb. at (-13.6) (9 gallons usable).		
Control Surface Movements	Elevator	Up 27 <sup>o</sup>	Down 17 <sup>o</sup>
	Elevator Tab	Up 12 <sup>o</sup>	Down 18 <sup>o</sup>
	Rudder	Left 24 <sup>o</sup>	Right 24 <sup>o</sup>
	Aileron	Up 21 <sup>o</sup>	Down 17 <sup>o</sup>
Serial Numbers (S/N) Eligible	600-1311D and subsequent		
Equipment	<p>The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. In addition, the following equipment is required:</p> <p>(a) FAA approved flight manual, dated November 1, 1965, or later FAA approved revision.</p> <p>(b) Operative pre-stall warning system per Snow Dwg. No. 90096.</p> <p>Either or both of the following items may be installed as customer optional equipment:</p> <p>(a) Canopy installation, Snow Dwg. No. 10131.</p> <p>(b) 12 or 24 volt electrical system, Snow Dwg. No. 90111.</p>		
Agricultural Dispersal Equipment	<p>Any one of the ten following agricultural dispersal systems may be installed:</p> <p>(a) 2" External Spray Installation, Snow Dwg. No. 80185.</p> <p>(b) 1-1/4" Internal Spray Installation, Snow Dwg. No. 80186.</p> <p>(c) Small Swathmaster Dispersal Installation, Snow Dwg. No. 80187 (See NOTE 2(n) for required placard).</p> <p>(d) Snow Spreader Dispersal Installation, Snow Dwg. No. 80188 (See NOTE 2(n) for required placard).</p> <p>(e) Quick Disconnect Flange and Snow Spreader Installation, Snow Dwg. No. 80609.</p>		

## Agricultural Dispersal Equipment (cont'd)

- (f) Large Swathmaster Installation - Standard of Swedish Gates, Snow Dwg. No. 80610.
- (g) Large Swathmaster Installation - 6 inch Adapter Box, Snow Dwg. No. 80602.
- (h) Spray System Installation - Fire Bomber Hopper with Cast Door, Snow Dwg. No. 80602.
- (i) Cable Dump System - Swedish Gate, Snow Dwg. No. 80251.
- (j) Fire Bomber Installation and Hopper Modification, Snow Dwg. No. 5-8062, Rev. D.

**II-Model S2R (S-2R), 1 PCLM (Restricted Category Only), Approved March 21, 1968**

Engine Pratt & Whitney WASP R-1340-AN-1 (S3H1 or S1H1 Commercial designation) with carburetor parts list settings 395118-3 or A-18639-7. Manifold pressure gage is to be modified per Drawing 60600 when the S1H1 engine is used. (See NOTE 5 for optional engine installation)

Fuel 80/87 minimum grade aviation gasoline

## Engine Limits

	<u>S3H1</u>				<u>S1H1</u>	
	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>
Takeoff (5 min.)	600	2,250	36.0	S.L.	36.5	S.L.
Max. Continuous	550	2,200	34.0	S.L.	35.0	S.L.
Max. Continuous	550	2,200	32.5	5,000	33.0	8,000

Propeller and Propeller Limits Hamilton Standard, constant speed, 12 D40 hub, 6101-12 blades. Diameter 109 inches maximum, 107 inches minimum. Pitch settings 11.5<sup>0</sup> low and 27.0<sup>0</sup> high at 42 inch station. Alternate settings, 11.5<sup>0</sup> low and 21.5<sup>0</sup> high at 42 inch station. Alternate blades, EAC AG100-2 - Diameter 106 inches (2 percent cutoff permitted). Pitch setting, 11.5<sup>0</sup> low and 20<sup>0</sup> high at 42 inches.

Airspeed Limits (CAS) (See Notes 2(o), 2(p), and 2(q) for exceptions)	Vne (Never Exceed)	159 m.p.h. (138 knots)
	Vp (Maneuvering)	126 m.p.h. (109 knots)
	Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)
	Vfe (Flap Extended)	123 m.p.h. (107 knots)

C.G. Range (+22.5) to (+30.0)

Maximum Weight 6,000 lbs.

Number of Seats 1 (+89.0)

Maximum Cargo Load See weight and balance data. Maximum baggage compartment, 60 lbs. (+112). Maximum hopper load, 3,336 lbs. (+29.9).

Fuel Capacity S/N 1380R - 70 gallons (38.5) (66 gallons usable capacity, one 35 gallon tank in each wing, tanks interconnected).  
S/N 1416R and subsequent - 106 gallons (38.5).  
S/N 1416R thru 1418R - (100 gallon usable capacity, one 53 gallon tank in each wing, tanks interconnected).  
S/N 1419R thru 1499R and subsequent and S/N 1501R thru 1510R - (98 gallon usable, one 53 gallon tank in each wing, tanks interconnected).  
S/N 1500R, 1511R and subsequent - (104 gallon usable, one 53 gallon tank in each wing, tanks interconnected).

Fuel Capacity (cont'd)	See NOTE 1 for data on unusable fuel. Also see NOTE 9 for other approved fuel capacities.															
Oil Capacity	11.4 gallons total (84 lbs. at -13.6) (9 gallons usable).															
Control Surface Movements	<table border="0"> <tr> <td>Elevator</td> <td>Up <math>27^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Elevator Tab</td> <td>Up <math>13^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>18^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Rudder</td> <td>Left <math>24^{\circ} \pm 1^{\circ}</math></td> <td>Right <math>24^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Aileron</td> <td>Up <math>21^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Flaps</td> <td></td> <td>Down <math>26^{\circ} - 30^{\circ}</math></td> </tr> </table>	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Flaps		Down $26^{\circ} - 30^{\circ}$
Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$														
Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$														
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Flaps		Down $26^{\circ} - 30^{\circ}$														
Serial Numbers Eligible	1380R, 1416R thru 4999R															
Equipment	<p>The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. In addition, the following equipment is required:</p> <ul style="list-style-type: none"> <li>(a) Airplane Flight Manual, dated February 27, 1979, and Supplement for Restricted Category Operation, dated February 27, 1979, or later FAA approved revision. (Only required for S/N 2526R and up).</li> <li>(b) 24 volt electrical system, Rockwell Drawing 90159. (24 volt system includes required wing night lights), effective S/N 1380R, 1416R thru 1590R.</li> <li>(c) 24 volt electrical system, Rockwell Drawing 90326, effective S/N 1591R and subsequent.</li> <li>(d) Operative pre-stall warning system per Rockwell Drawing 90095, S/N 1416R thru 1440R.</li> </ul>															
Agricultural Dispersal Equipment	<p>Any one of the following agricultural dispersal systems may be installed with the R1340 engines, or with the optional Wright R-1300-1B engine installation:</p> <ul style="list-style-type: none"> <li>(a) 2" External Spray Installation, Aero Commander Dwg. No. 80680, S/N 1416R thru 1510R.</li> <li>(b) Spreader and Calibration Installation, Aero Commander Dwg. No. 80674, S/N 1416R and subsequent.</li> <li>(c) Fire Bomber Dump System Installation, Aero Commander Dwg. No. 80792 (See NOTE 2(o) for required placard), S/N 1416R thru 1576R.</li> <li>(d) Micronair Spray System, Aero Commander Dwg. No. 80870 (See NOTE 2(q) for required placard), S/N 1416R and subsequent.</li> <li>(e) 2" Low Drag Spray System, Aero Commander Dwg. No. 81012, S/N 1511R thru 1620R.</li> <li>(f) Boommaster Installation, Aero Commander Dwg. No. 80931, S/N 1416R and subsequent.</li> <li>(g) Standard Spray System, Rockwell Dwg. No. 81071, S/N 1621R and subsequent.</li> <li>(h) Spreader and Spreader Quick-Disconnect Installation, Rockwell Dwg. No. 80975, S/N 1416R and subsequent.</li> <li>(i) Large Swathmaster - Small Gate Installation, Rockwell Dwg. No. 80815, S/N 1416R thru 2068R.</li> </ul>															

## Agricultural Dispersal Equipment (cont'd)

- (j) Swathmaster Installation, Rockwell Dwg. No. 81061, S/N 1416R thru 2068R.
- (k) 2" Spray System Installation, Rockwell Dwg. No. 80852, S/N 1511R thru 1620R.
- (l) Spray System Installation, Rockwell Dwg. No. 80854, S/N 1511R and subsequent.
- (m) Fire Bomber System Installation, Rockwell Dwg. No. 81069, S/N 1577R and subsequent.

**III-Model S2R-T34, 1 PCLM (Restricted Category Only), Approved April 28, 1977**

See Note 8 for two-place configuration. See Notes 18 and 19 for options. See Note 22 for life limited parts.

## Engine

Pratt & Whitney (United Aircraft of) Canada PT6A-34AG

Optional Engines: Pratt & Whitney Canada PT6A-34 (See NOTE 12 for instructions),

Pratt & Whitney Canada PT6A-36 (Dry Configuration Only),

Pratt & Whitney Canada PT6A-41, PT6A-41AG, and PT6A-42 (See NOTE 14 for more information on airplanes with these optional engines)

## Fuel

Jet A, Jet B, JP-4, JP-5, Automotive Diesel Number 1D or 2D in accordance with ACL Service Bulletin Number 1344. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.) Automotive diesel fuel is approved only for agricultural application flights and only when the free air temperature is above:

+20°F for Grade No. 1D

+40°F for Grade No. 2D

## Oil

UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.

## Engine Limits

PT6A-34AG/-34/-36:

	Takeoff and <u>Max. Cont.</u>	Transient <u>Start/Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	750			
Torque (PSI)	64.5	68.4 Trans (2 sec.)	64.5	
ITT (°C)	790	1,090 Start (2 sec.)	790	
Ng (%)	101.5	102.6 Trans (2 sec.)	101.5	
Np (RPM)	2,200	2,420 Trans (2 sec.)	2,100	
Oil Press (PSIG)	85 to 105	85 to 105	85 to 105	40 min.
Oil Temp (°C)	10 to 99	-40 minimum	0 to 99	-40 to 99

The ratings shown are based on the static sea level standard condition with no external accessory loads and no air bleed.

## Propeller and Propeller Limits

(See Note 24 for pitch limits)

Hartzell Hub Model HC-B3TN-3C (or HC-B3TN-3D) with Blade Model T-10282, Diameter 102.5 inches maximum, 92.5 inches minimum, or Optional Blade Model T-10282(N)+4, Diameter 106 inches maximum, 98 inches minimum.

## Airspeed Limits (CAS)

Vne (Never Exceed)	159 mph (138 knots)
Vp (Maneuvering)	126 mph (109 knots)
Vno (Max. Structural Cruising)	126 mph (109 knots)
Vfe (Flap Extended)	123 mph (107 knots)

## C. G. Range

(See Note 8 for two-place)

Forward limit at 6,000 lbs., +26.5 inches aft of datum.

Forward limit at 4,000 lbs., and below +24.0 inches aft of datum.

(Straight line variation in the forward limit between 4,000 and 6,000 lbs.)

Aft limit +30.0 inches aft of datum.

Datum is the leading edge of the wing.

Maximum Weight	6,000 lbs.		
Maximum Operating Altitude	12,000 feet		
Number of Seats	1 (+89) (see NOTE 8 for two-place).		
Maximum Cargo Load	See weight and balance data. Maximum baggage compartment, 60 lbs. (+112). (See NOTE 8 for two-place). Maximum hopper load, 3,336 lbs. (+29.9). (See NOTE 10 for increased load limit).		
Fuel Capacity	104 gallons usable, one 53 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel. Also see NOTE 9 for other approved fuel capacities.		
Oil Tank Capacity	11 quarts - usable oil tank capacity 6 quarts.		
Control Surface Movements	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$
	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible (See Note 8 for two-place)	6000 - 6049 T34-001 and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved June 23, 1978, and Supplement for Restricted Category Operation approved June 23, 1978, or later approved versions.		
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems. CAUTION: For operation with the Micronair Spray Equipment System or the Fire Bomber System, or with any system when an Agavenco pump is installed, the placards for airspeed limitations referred to in NOTE 2(q), 2(o), or 2(p), respectively, for the S2R are applicable.		

**IV-Model S2R-T15, 1 PCLM (Restricted Category Only), Approved April 3, 1979**

See Note 8 for two-place configuration. See Notes 18 and 19 for options. See Note 22 for life limited parts.

Engine	Pratt & Whitney (United Aircraft of) Canada PT6A-15AG or PT6A-27
	Due to anticipated operating environment, servicing and overhaul interval for both the PT6A-15AG and PT6A-27 engines shall be in accordance with Pratt & Whitney's recommendations for the PT6A-15AG engine.
Fuel	Jet A, Jet B, JP-4, JP-5, Automotive Diesel Number 1D or 2D in accordance with UACL Service Bulletin Number 1344. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.) Automotive diesel fuel is approved only for agricultural application flights and only when the free air temperature is above: +20°F for Grade No. 1D +40°F for Grade No. 2D
Oil	UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.

## Engine Limits

	Takeoff and <u>Max. Cont.</u>	Transient <u>Start/Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	680			
Torque (PSI)	53.0	68.8 Trans (2 sec.)	53.0	
ITT (°C)	725	1,090 Start (2 sec.)	725	
Ng (%)	101.5	102.7 Trans (10 sec.)	101.5	
Np (RPM)	2,200	2,420 Trans (10 sec.)	2,100	
Oil Press (PSIG)	80 to 100	80 to 100	80 to 100	40 min.
Oil Temp (°C)	10 to 99	-40 minimum	0 to 99	-40 to 99

The ratings shown are based on the static sea level standard condition with no external accessory loads and no air bleed.

Propeller and Propeller Limits  
(See Note 24 for pitch limits)

Hartzell Hub Model HC-B3TN-3C (or HC-B3TN-3D) with Blade Model T-10282, Diameter 102.5 inches maximum, 92.5 inches minimum or optional Blade Model T-10282(N)+4, Diameter 106 inches maximum, 98 inches minimum.

## Airspeed Limits (CAS)

Vne (Never Exceed)	159 mph (138 knots)
Vp (Maneuvering)	126 mph (109 knots)
Vno (Max. Structural Cruising)	126 mph (109 knots)
Vfe (Flap Extended)	123 mph (107 knots)

C. G. Range  
(See Note 8 for two-place)

Forward limit at 6,000 lbs., +26.5 inches aft of datum.  
Forward limit at 4,000 lbs. and below, +24.0 inches aft of datum.  
(Straight line variation in the forward limit between 4,000 and 6,000 lbs.)  
Aft limit +30.0 inches aft of datum.  
Datum is the leading edge of the wing.

## Maximum Weight

6,000 lbs.

## Maximum Operating Altitude

12,000 feet

## Number of Seats

1 (+89) (see NOTE 8 for two-place)

## Maximum Cargo Load

See weight and balance data.  
Maximum baggage compartment, 60 lbs. (+112). (See NOTE 8 for two-place).  
Maximum hopper load, 3,336 lbs. (+29.9). (See NOTE 10 for increased load limit).

## Fuel Capacity

104 gallons usable, one 53 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel. Also see NOTE 9 for other approved fuel capacities.

## Oil Tank Capacity

11 quarts - usable oil tank capacity 6 quarts.

## Control Surface Movements

Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$
Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
Flaps		Down $15^{\circ} \pm 1^{\circ}$

Serial Numbers Eligible  
(See Note 8 for two-place)

T15-001 and subsequent

## Required Equipment

The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved April 3, 1979, and Supplement for Restricted Category Operation approved April 3, 1979, or later approved versions.

Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems. CAUTION: For operation with the Micronair Spray Equipment System or the Fire Bomber System, or with any system when an Agavenco pump is installed, the placards for airspeed limitations referred to in NOTE 2(q), 2(o), or 2(p), respectively, for the S2R are applicable.
----------------------------------	---

**V-Model S2R-R3S, 1 PCLM (Restricted Category Only), Approved August 1, 1979**

See Note 8 for two place configuration.

Engine	Wsk - "Pezetel" PZL-3S															
Fuel	100/130 Minimum grade aviation gasoline															
Oil	Aeroshell 100 or equivalent															
Engine Limits	<table> <thead> <tr> <th></th> <th><u>H.P.</u></th> <th><u>R.P.M.</u></th> <th><u>M.P.(In. Hg.)</u></th> <th><u>ALT.</u></th> </tr> </thead> <tbody> <tr> <td>Takeoff (1 min.)</td> <td>592</td> <td>2,200</td> <td>37.0</td> <td>S.L.</td> </tr> <tr> <td>Max. Continuous</td> <td>594</td> <td>2,050</td> <td>36.2</td> <td>S.L.</td> </tr> </tbody> </table>		<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>	Takeoff (1 min.)	592	2,200	37.0	S.L.	Max. Continuous	594	2,050	36.2	S.L.
	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>												
Takeoff (1 min.)	592	2,200	37.0	S.L.												
Max. Continuous	594	2,050	36.2	S.L.												
Propeller and Propeller Limits	One Dowty Rotol, Ltd., Model (C) R. 289/3-110-F/1, Constant Speed, Hydraulic, Non-Feathering, Non-Reversing Pitch Control with Pezetel Governor 0719-812008. Blade Model 660705200, Diameter: 102" $\pm$ 0.0 Pitch Setting at 37" Blade Radius Low $12^{\circ} \pm 1/4^{\circ}$ ; High $20^{\circ} \pm 1/4^{\circ}$ .															
or	One WSK Model US-132000/A Hub, US-132500 Blades, Diameter 103.15 inches maximum, 102.0 minimum, Pitch Setting at 37" Blade Radius Low $12^{\circ} \pm 1/2^{\circ}$ ; High $32^{\circ} \pm 1^{\circ}$ .															
Airspeed Limits (CAS)	<table> <tbody> <tr> <td>Vne (Never Exceed)</td> <td>159 mph (138 knots)</td> </tr> <tr> <td>Vp (Maneuvering)</td> <td>126 mph (109 knots)</td> </tr> <tr> <td>Vno (Max. Structural cruising)</td> <td>126 mph (109 knots)</td> </tr> <tr> <td>Vfe (Flap Extended)</td> <td>123 mph (107 knots)</td> </tr> </tbody> </table>	Vne (Never Exceed)	159 mph (138 knots)	Vp (Maneuvering)	126 mph (109 knots)	Vno (Max. Structural cruising)	126 mph (109 knots)	Vfe (Flap Extended)	123 mph (107 knots)							
Vne (Never Exceed)	159 mph (138 knots)															
Vp (Maneuvering)	126 mph (109 knots)															
Vno (Max. Structural cruising)	126 mph (109 knots)															
Vfe (Flap Extended)	123 mph (107 knots)															
C. G. Range	(+22.5) to (+27.5) (See NOTE 8 for two-place)															
Maximum Weight	6,000 lbs.															
Number of Seats	1 (+89) (See NOTE 8 for two-place)															
Maximum Cargo Load	See weight and balance data. Maximum baggage compartment, 60 lbs. (+112). (See NOTE 8 for two-place). Maximum hopper load, 3,336 lbs. (+29.9).															
Fuel Capacity	S/N R3S-001 and subsequent - (104 gallons usable, one 53 gallon tank in each wing, tanks interconnected). See NOTE 1 for data on unusable fuel.															
Oil Tank Capacity	11.4 gallons total (84 lbs. at -13.6) (9.0 gallons usable).															
Control Surface Movements	<table> <tbody> <tr> <td>Elevator</td> <td>Up <math>27^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Elevator Tab</td> <td>Up <math>13^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>18^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Rudder</td> <td>Left <math>24^{\circ} \pm 1^{\circ}</math></td> <td>Right <math>24^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Aileron</td> <td>Up <math>21^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Flaps</td> <td></td> <td>Down <math>26^{\circ} - 30^{\circ}</math></td> </tr> </tbody> </table>	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Flaps		Down $26^{\circ} - 30^{\circ}$
Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$														
Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$														
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Flaps		Down $26^{\circ} - 30^{\circ}$														
Serial Numbers Eligible	R3S-001 and subsequent. (See NOTE 8 for two-place)															
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved August 1, 1979, or later approved versions.															

Agricultural Dispersal Equipment See NOTE 17 for dispersal systems.

**VI-Model S2R-T11, 1 PCLM (Restricted Category Only), Approved October 26, 1979**

See Notes 8 for two-place configuration. See Notes 18 and 19 for options. See Note 22 for life limited parts.

Engine Pratt & Whitney (United Aircraft of) Canada PT6A-11AG

Fuel Jet A, Jet B, JP-4, JP-5, Automotive Diesel Number 1D or 2D in accordance with UACL Service Bulletin Number 1344. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.) Automotive diesel fuel is approved only for agricultural application flights and only when the free air temperature is above:  
 +20°F for Grade No. 1D  
 +40°F for Grade No. 2D

Oil UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.

Engine Limits

	<u>Takeoff and Max. Cont.</u>	<u>Transient Start/Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	500			
Torque (PSI)	38.6	48.5 Trans (2 sec.)	38.6	
ITT (°C)	700	1,090 Start (2 sec.)	700	
Ng (%)	101.5	102.6 Trans (10 sec.)	101.5	
Np (RPM)	2,200	2,420 Trans (10 sec.)	2,068	
Oil Press (PSIG)	80 to 100	80 to 100	80 to 100	40 min.
Oil Temp (°C)	10 to 99	-40 min.	0 to 99	-40 to 99

The ratings shown on the United Aircraft of Canada PT6A-11AG engine are based on the static sea level standard condition with no external accessory loads and no air bleed.

Propeller and Propeller Limits (See Note 24 for pitch limits) Hartzell Hub Model HC-B3TN-3C (or HC-B3TN-3D) with Blade Model T-10282, Diameter 102.5 inches maximum, 92.5 inches minimum or optional Blade Model T-10282(N)+4, Diameter 106 inches maximum, 98 inches minimum.

Airspeed Limits (CAS)

Vne (Never Exceed)	159 mph (138 knots)
Vp (Maneuvering)	126 mph (109 knots)
Vno (Max. Structural Cruising)	126 mph (109 knots)
Vfe (Flap Extended)	123 mph (107 knots)

C. G. Range (See Note 8 for two-place)

Forward limit at 6,000 lbs., +26.5 inches aft of datum.  
 Forward limit at 4,000 lbs. and below, +24.0 inches aft of datum.  
 (Straight line variation in the forward limit between 4,000 and 6,000 lbs.).  
 Aft limit +30.0 inches aft of datum.  
 Datum is the leading edge of the wing.

Maximum Weight 6,000 lbs.

Maximum Operating Altitude 12,000 feet

Number of Seats 1 (+89) (See NOTE 8 for two-place)

Maximum Cargo Load See weight and balance data.  
 Maximum baggage compartment, 60 lbs. (+112). (See NOTE 8 for two-place).  
 Maximum hopper load, 3,336 lbs. (+29.9). (See NOTE 10 for increased load limit).

Fuel Capacity	104 gallons usable, one 53 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel. Also see NOTE 9 for other approved fuel capacities.															
Oil Tank Capacity	11 quarts - usable oil tank capacity 6 quarts.															
Control Surface Movements (See Note 8 for two-place)	<table border="0"> <tr> <td>Elevator</td> <td>Up <math>27^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Elevator Tab</td> <td>Up <math>13^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>18^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Rudder</td> <td>Left <math>24^{\circ} \pm 1^{\circ}</math></td> <td>Right <math>24^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Aileron</td> <td>Up <math>21^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Flaps</td> <td></td> <td>Down <math>15^{\circ} \pm 1^{\circ}</math></td> </tr> </table>	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Elevator Tab	Up $13^{\circ} \pm 1^{\circ}$	Down $18^{\circ} \pm 1^{\circ}$														
Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$														
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Flaps		Down $15^{\circ} \pm 1^{\circ}$														
Serial Numbers Eligible	T11-001 and subsequent. (See NOTE 8 for two-place)															
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved October 26, 1979, and Supplement for Restricted Category Operation approved October 26, 1979, or later approved versions.															
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems. CAUTION: For operation with the Micronair Spray Equipment System or the Fire Bomber System, or with any system when an Agavenco pump is installed, the placards for airspeed limitations referred to in NOTE 2(q), 2(o), or 2(p), respectively, for the S2R are applicable.															

**VII-Model S2R-R1340, 2 PCLM (Restricted Category Only)\*, Approved May 6, 1980**

\*See Note under certification basis for approved special purpose operations. Also see Note 15 for one-place configuration.

Engine	Pratt & Whitney WASP R-1340-AN-1 (S3H1 or S1H1 Commercial designation) with carburetor parts list settings 395118-3 or A-18639-7. Manifold pressure gage is to be modified per Drawing 60600 when the S1H1 engine is used.																																												
Fuel	80/87 minimum grade aviation gasoline																																												
Engine Limits	<table border="0"> <thead> <tr> <th></th> <th colspan="4">S3H1</th> <th colspan="3">S1H1</th> </tr> <tr> <th></th> <th><u>H.P.</u></th> <th><u>R.P.M.</u></th> <th><u>M.P.In.</u></th> <th><u>H.G.</u></th> <th><u>ALT.</u></th> <th><u>M.P.In.</u></th> <th><u>H.G.</u></th> <th><u>ALT.</u></th> </tr> </thead> <tbody> <tr> <td>Takeoff (5 min.)</td> <td>600</td> <td>2,250</td> <td>36.0</td> <td>S.L.</td> <td>36.5</td> <td>S.L.</td> <td></td> <td></td> </tr> <tr> <td>Max. Continuous</td> <td>550</td> <td>2,200</td> <td>34.0</td> <td>S.L.</td> <td>35.0</td> <td>S.L.</td> <td></td> <td></td> </tr> <tr> <td>Max. Continuous</td> <td>550</td> <td>2,200</td> <td>32.5</td> <td>5,000</td> <td>33.0</td> <td>8,000</td> <td></td> <td></td> </tr> </tbody> </table>		S3H1				S1H1				<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.In.</u>	<u>H.G.</u>	<u>ALT.</u>	<u>M.P.In.</u>	<u>H.G.</u>	<u>ALT.</u>	Takeoff (5 min.)	600	2,250	36.0	S.L.	36.5	S.L.			Max. Continuous	550	2,200	34.0	S.L.	35.0	S.L.			Max. Continuous	550	2,200	32.5	5,000	33.0	8,000		
	S3H1				S1H1																																								
	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.In.</u>	<u>H.G.</u>	<u>ALT.</u>	<u>M.P.In.</u>	<u>H.G.</u>	<u>ALT.</u>																																					
Takeoff (5 min.)	600	2,250	36.0	S.L.	36.5	S.L.																																							
Max. Continuous	550	2,200	34.0	S.L.	35.0	S.L.																																							
Max. Continuous	550	2,200	32.5	5,000	33.0	8,000																																							
Propeller and Propeller Limits	<p>Hamilton Standard, constant speed, 12D40 Hub, 6101-12 blades. Diameter 109 inches maximum, 107 inches minimum. Pitch settings, <math>11.5^{\circ}</math> low and <math>27.0^{\circ}</math> high at 42 inch station. Alternate settings, <math>11.5^{\circ}</math> low and <math>21.5^{\circ}</math> high at 42 inch station.</p> <p>Alternate blades, EAC AG100-2, Diameter 106 inches (2 percent cutoff permitted). Pitch settings <math>11.5^{\circ}</math> low and <math>20^{\circ}</math> high at 42 inches.</p>																																												
Airspeed Limits (CAS)	<table border="0"> <tr> <td>Vne (Never Exceed)</td> <td>159 m.p.h. (138 knots)</td> </tr> <tr> <td>Vp (Maneuvering)</td> <td>126 m.p.h. (109 knots)</td> </tr> <tr> <td>Vno (Max. Structural Cruising)</td> <td>126 m.p.h. (109 knots)</td> </tr> <tr> <td>Vfe (Flap Extended)</td> <td>123 m.p.h. (107 knots)</td> </tr> </table>	Vne (Never Exceed)	159 m.p.h. (138 knots)	Vp (Maneuvering)	126 m.p.h. (109 knots)	Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)	Vfe (Flap Extended)	123 m.p.h. (107 knots)																																				
Vne (Never Exceed)	159 m.p.h. (138 knots)																																												
Vp (Maneuvering)	126 m.p.h. (109 knots)																																												
Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)																																												
Vfe (Flap Extended)	123 m.p.h. (107 knots)																																												
C.G. Range	(+22.5) to (+30.0) with Elevator Down Spring, P/N 19661-1, and for S/N R1340-036DC and up (+22.5) to (+27.5) without P/N 19661-1 installed.																																												
Maximum Weight	6,000 lbs.																																												

Number of Seats	1 (+89.0) (See NOTE 15 for one-place configuration) 1 (+127 - Forward Facing) or (+111 - Aft Facing)															
Maximum Cargo Load	See weight and balance data. Maximum passenger/cargo compartment, 200 lbs. (+120) (See NOTE 15 for one-place configuration) Maximum hopper load, 3,336 lbs. (+29.9).															
Fuel Capacity	S/N R1340-001DC to R1340-035: 104 gallons usable, one 53 gallon tank in each wing, tanks interconnected; S/N R1340-001DC and up: 133 gallons usable, one 68 gallon tank in each wing. See NOTE 1 for data on unusable fuel. See NOTE 9 for other approved fuel capacities.															
Oil Tank Capacity	11.4 gallon total (84 lbs. at -13.6) (9.0 gallons usable).															
Control Surface Movements	<table border="0"> <tr> <td>Elevator</td> <td>Up <math>27^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Elevator Tab</td> <td>Up <math>8^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>22^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Rudder</td> <td>Left <math>24^{\circ} \pm 1^{\circ}</math></td> <td>Right <math>24^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Aileron</td> <td>Up <math>21^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Flaps</td> <td></td> <td>Down <math>15^{\circ} \pm 1^{\circ}</math></td> </tr> </table>	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$														
Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$														
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Flaps		Down $15^{\circ} \pm 1^{\circ}$														
Serial Numbers Eligible	R1340-001DC and subsequent See NOTE 15 for S/N R1340-011 and subsequent (one-place configuration)															
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include for S/N R1340-001DC to S/N R1340-035 the Airplane Flight Manual and Supplement for Restricted Category Operation, dated May 6, 1980, or later approved versions; or for S/N R1340-036 and up, the Thrush Airplane Flight Manual, dated November 29, 2007, or later approved versions.															
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems.															
Structural Limitations	For S/N R1340-036DC and up (with or without DC suffix) wing lower spar cap p/n's 20207-015 and 20207-016 must be replaced at 28,800 hours time in service and tail landing gear spring p/n 5079-1 must be replaced every 5,000 hours time in service.  For S/N R1340-036DC and up (with or without DC suffix), mandatory inspections in accordance with the Airworthiness Limitations contained in Thrush Aircraft Maintenance Manual for the S2R-R1340 effective January 1, 2008, or later FAA accepted revision, are required.															

**VIII-Model S2R-R1820, 2 PCLM (Restricted Category Only)\*, Approved February 20, 1981**

\*See Note under certification basis for approved special purpose operations. Also see Note 15 for one-place configuration.

Engine	Wright R-1820-71, -60, -97, -99; GR-1820G-202A; 702C9GC1, 2, 3, 4; 704C9GC1, 2, 3, 4			
Fuel	100/130 minimum grade aviation gasoline			
Engine Limits	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>
Takeoff (1 min.)	1200	2,500	45.5	S.L.
Max. Continuous	1000	2,300	39.5	S.L.
Max. Continuous	1000	2,300	37.2	6,900
	Straight line variation between points given.			

Propeller and Propeller Limits	<p>a. With Wright R-1820-71, -60, GR-1820G-202A; 702C9GC1, 2, 3, 4; 704C9GC1, 2, 3, 4 engines: Hamilton Standard 43D50 or 33D50 constant speed -</p> <ol style="list-style-type: none"> <li>1. Hub Model: 43D50-321 Blade Model: 6933A-9 Diameter: 111" Maximum, 109" Minimum Pitch Settings: At 42 inch Station - Low 21.5<sup>o</sup>, High 52.5<sup>o</sup></li> <li>2. Hub Model: 33D50-119 Blade Model: 6601-18S or 7005-18S Diameter: 120-5/8" Maximum, 117-5/8" Minimum Pitch Settings: At 42 inch Station -Low 19<sup>o</sup>, High 39<sup>o</sup></li> </ol> <p>b. With Wright R-1820-97 or R-1820-99 engines: Hamilton Standard 23E50 constant speed -</p> <p>Blade Model: Serv-Aero SA10P-18Q Diameter: 120-1/4" Maximum, 117-3/4" Minimum Pitch Settings: At 42 inch Station - Low 26<sup>o</sup>, High 50<sup>o</sup></p> <p>c. Governor: Hamilton Standard 4G-10-7</p>															
Airspeed Limits (CAS)	<table border="0"> <tr> <td>Vne (Never Exceed)</td> <td>159 m.p.h. (138 knots)</td> </tr> <tr> <td>Vp (Maneuvering)</td> <td>126 m.p.h. (109 knots)</td> </tr> <tr> <td>Vno (Max. Structural Cruising)</td> <td>126 m.p.h. (109 knots)</td> </tr> <tr> <td>Vfe (Flap Extended)</td> <td>123 m.p.h. (107 knots)</td> </tr> </table>	Vne (Never Exceed)	159 m.p.h. (138 knots)	Vp (Maneuvering)	126 m.p.h. (109 knots)	Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)	Vfe (Flap Extended)	123 m.p.h. (107 knots)							
Vne (Never Exceed)	159 m.p.h. (138 knots)															
Vp (Maneuvering)	126 m.p.h. (109 knots)															
Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)															
Vfe (Flap Extended)	123 m.p.h. (107 knots)															
C.G. Range (See Note 15 for one-place)	<p>(+23) to (+30.0) with Elevator Down Spring, P/N 19661-1 (+23) to (+27.5) without P/N 19661-1 installed.</p>															
Maximum Weight	6,000 lbs.															
Number of Seats	1 (+89.0) (See NOTE 15 for one-place configuration) 1 (+127 - Forward Facing) or (+111 - Aft Facing)															
Maximum Cargo Load	<p>See weight and balance data. Maximum passenger/cargo compartment, 200 lbs. (+120). See NOTE 15 for one-place configuration. Maximum hopper load, 3,336 lbs. (+29.9). See NOTE 10 for increased load limit.</p>															
Fuel Capacity	S/N R1820-001DC and subsequent - (190 gallons usable, one 96 gallon tank in each wing, tanks interconnected). See NOTE 1 for data on unusable fuel. Also see NOTE 9 for other approved fuel capacities.															
Oil Tank Capacity	13 gallons total at Station (+153). See NOTE 15 for one-place configuration.															
Control Surface Movements	<table border="0"> <tr> <td>Elevator</td> <td>Up 27<sup>o</sup> ± 1<sup>o</sup></td> <td>Down 17<sup>o</sup> ± 1<sup>o</sup></td> </tr> <tr> <td>Elevator Tab</td> <td>Up 12<sup>o</sup> ± 1<sup>o</sup></td> <td>Down 22<sup>o</sup> ± 1<sup>o</sup></td> </tr> <tr> <td>Rudder</td> <td>Left 24<sup>o</sup> ± 1<sup>o</sup></td> <td>Right 24<sup>o</sup> ± 1<sup>o</sup></td> </tr> <tr> <td>Aileron</td> <td>Up 21<sup>o</sup> ± 1<sup>o</sup></td> <td>Down 17<sup>o</sup> ± 1<sup>o</sup></td> </tr> <tr> <td>Flaps</td> <td></td> <td>Down 15<sup>o</sup> ± 1<sup>o</sup></td> </tr> </table>	Elevator	Up 27 <sup>o</sup> ± 1 <sup>o</sup>	Down 17 <sup>o</sup> ± 1 <sup>o</sup>	Elevator Tab	Up 12 <sup>o</sup> ± 1 <sup>o</sup>	Down 22 <sup>o</sup> ± 1 <sup>o</sup>	Rudder	Left 24 <sup>o</sup> ± 1 <sup>o</sup>	Right 24 <sup>o</sup> ± 1 <sup>o</sup>	Aileron	Up 21 <sup>o</sup> ± 1 <sup>o</sup>	Down 17 <sup>o</sup> ± 1 <sup>o</sup>	Flaps		Down 15 <sup>o</sup> ± 1 <sup>o</sup>
Elevator	Up 27 <sup>o</sup> ± 1 <sup>o</sup>	Down 17 <sup>o</sup> ± 1 <sup>o</sup>														
Elevator Tab	Up 12 <sup>o</sup> ± 1 <sup>o</sup>	Down 22 <sup>o</sup> ± 1 <sup>o</sup>														
Rudder	Left 24 <sup>o</sup> ± 1 <sup>o</sup>	Right 24 <sup>o</sup> ± 1 <sup>o</sup>														
Aileron	Up 21 <sup>o</sup> ± 1 <sup>o</sup>	Down 17 <sup>o</sup> ± 1 <sup>o</sup>														
Flaps		Down 15 <sup>o</sup> ± 1 <sup>o</sup>														
Serial Numbers Eligible	R1820-001DC and subsequent															
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved February 20, 1981, or later approved version.															
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems.															

**IX-Model S2R-T65, 2 PCLM (Restricted Category Only)\*, Approved September 3, 1987**

\*See Note under certification basis for approved special purpose operations. See Note 22 for life limited parts.

Engine	Pratt & Whitney (United Aircraft of) Canada PT6A-65AG
Fuel	Jet A, Jet B, JP-4, JP-5, in accordance with P&WC Service Bulletin Number 13244. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.)
Oil	UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.

## Engine Limits

	Takeoff and <u>Max. Cont.</u>	Transient <u>Start/Accel.</u>	<u>Idle</u>
SHP	1,230		
Torque (PSI) (2 sec.)	45.4	61.0 Trans	
ITT (°C)	810	1,000 Start (5 sec.)	715
Ng (%)	104		58
Np (RPM)	1,700	1,870 Trans (5 sec.)	
Oil Press (PSIG)	90 to 135		60 min.
Oil Temp (°C)	0 to 110	0 to 110	-40 to 110

Propeller and Propeller Limits (See Note 24 for pitch limits)	Hartzell HC-B5MP-3C propeller, constant speed, feathering and reversing; Hub Model HC-B5MP-3C; Blade Model M10876ANS. Diameter 111.0 maximum, 110.7 inches minimum.
--	---

## Certification Basis

- (1) CAR 8.10(a)(1), dated October 11, 1950, including the Airworthiness requirements of Appendix B.
- (2) 14 CFR Part 23, effective February 1, 1965, only as applicable to turboprop engine installations and listed by Part 23 section below.  
(The Part 23 amendment level is shown in parentheses.)

23.49(e)(2)(-21)	23.933(-7)	23.1045(-7)	23.1305(-15)
23.65(c)(-21)	23.937(-7)	23.1091(-7)	23.1323(-7)
23.75(b)(-7)	23.951(-15)	23.1093(-15)	23.1337(-7)
23.77(b)(-21)	23.955(-7)	23.1103(-7)	23.1353(-20)
23.173(-14)	23.959(-7)	23.1105(0)	23.1521(0)
23.175(-14)	23.977(-17)	23.1111(-7)	23.1527(-7)
23.177(0)	23.991(-7)	23.1121(-7)	23.1529(-8)
23.371(-7)	23.997(-15)	23.1141(-14)	23.1545(-7)
23.629(e)(-31)	23.1013(-15)	23.1143(-7)	23.1549(-17)
23.831(0)	23.1015(-15)	23.1145(-18)	23.1557(-14)
23.901(-7)	23.1019(-15)	23.1155(-7)	23.1583(-10)
23.903(-14)	23.1027(-14)	23.1165(0)	23.1587(a)(-7)
23.905(0)	23.1041(-7)	23.1183(-15)	
23.929(-14)	23.1043(-7)	23.1303(0)	

- (3) The intent of §25.305(c) regarding the dynamic response of the engine mount structure.

Airspeed Limits (CAS)	Vne (Never Exceed)	159 m.p.h. (138 knots)
	Vp (Maneuvering)	126 m.p.h. (109 knots)
	Vno (Max. Structural Cruising)	126 m.p.h. (109 knots)
	Vfe (Flap Extended)	123 m.p.h. (107 knots)

C.G. Range	Forward Limit +22.5 inches aft of datum Aft Limit +29.0 inches aft of datum. Datum is the leading edge of the wing.															
Maximum Weight	6,000 lbs.															
Maximum Operating Altitude	12,000 feet															
Number of Seats	1 (+89) 1 (+127)															
Maximum Cargo Load	See weight and balance data. Maximum cargo compartment, 200 lbs. (+120). Maximum hopper load, 4,000 lbs. (+29.9).															
Fuel Capacity	228 gallon usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.															
Oil Tank Capacity	11 quarts - usable oil tank capacity 6 quarts.															
Control Surface Movements (See Note 8 for two-place)	<table border="0"> <tr> <td>Elevator</td> <td>Up <math>27^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Elevator Tab</td> <td>Up <math>8^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>22^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Rudder</td> <td>Left <math>24^{\circ} \pm 1^{\circ}</math></td> <td>Right <math>24^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Aileron</td> <td>Up <math>21^{\circ} \pm 1^{\circ}</math></td> <td>Down <math>17^{\circ} \pm 1^{\circ}</math></td> </tr> <tr> <td>Flaps</td> <td></td> <td>Down <math>15^{\circ} \pm 1^{\circ}</math></td> </tr> </table>	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$														
Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$														
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$														
Flaps		Down $15^{\circ} \pm 1^{\circ}$														
Serial Numbers Eligible	T65-001DC and subsequent															
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved September 3, 1987, and Supplement for Restricted Category Operation approved September 3, 1987, or later approved version.															
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems. CAUTION: For operation with the Micronair Spray Equipment System or the Fire Bomber System, or with any system when an Agavenco pump is installed, the placards for airspeed limitations referred to in NOTE 2(q), 2(o), or 2(p), respectively, for the S2R are applicable.															
<b><u>X Model S2RHG-T65, 2 PCLM (Restricted Category Only)*, Approved June 8, 1988</u></b>																
*See Note under certification basis for approved special purpose operations. See NOTE 25 for single cockpit configuration with optional engines. See NOTE 22 for life limited parts.																
Engine	S/N T65-002DC thru T65-012DC: Pratt & Whitney Canada PT6A-65AG S/N T65HG-013DC and up: Pratt & Whitney Canada PT6A-60AG															
Fuel	Jet A, Jet B, JP-4, JP-5, in accordance with P&WC Service Bulletin Number 13244. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.)															
Oil	UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.															

Engine Limits	<u>PT6A-65AG:</u>		Takeoff and <u>Max. Cont.</u>	Transient <u>Start/Accel.</u>	<u>Idle</u>		
	SHP		1,230				
	Torque (PSI) (2 sec.)		45.4	61.0 Trans			
	ITT (°C)		810	1,000 Start (5 sec.)	715		
	Ng (%)		104		58		
	Np (RPM)		1,700	1,870 Trans (5 sec.)			
	Oil Press (PSIG)		90 to 135		60 min.		
	Oil Temp (°C)		0 to 110	0 to 110	-40 to 110		
	<u>PT6A-60AG:</u>		<u>Max.</u>	Transient*			
	<u>Takeoff</u>	<u>Cont.</u>	<u>Start</u>	<u>Accel.</u>	<u>Reverse</u>	<u>Idle</u>	
	SHP	1,050	1,020		900		
	Torque (PSI)**	38.8	37.7				
	ITT (°C)	820	775	1,000	850	760	750
	Ng (%)	104	104		104		58
	Np (RPM)	1700	1700		1,870	1,650	
	Oil Press (PSIG)	90 to 135	90 to 135	0 to 200	40 to 200	90 to 135	60 Minimum
	Oil Temp (°C)	10 to 110	0 to 110	-40 Min.	0 to 110	0 to 104	-40 to 110

\*Transient engine limits are 5 seconds for starting and 20 seconds for acceleration.

\*\*The Torque pressure limits listed are for NP=1,700 RPM only.

Propeller and Propeller Limits  
(See Note 24 for pitch limits)

Hartzell HC-B5MP-3C propeller, constant speed, feathering and reversing;  
Hub Model HC-B5MP-3C; Blade Model M10876AS or M10876ANS.  
Diameter 111.0 inches maximum, 110.7 inches minimum.

Certification Basis

- (1) 14 CFR Part 21.25(a)(1)
- (2) CAR 3, effective May 15, 1956, including Amendments 3-1 through 3-8 as modified by CAR 8.10(a)(1) effective October 11, 1950.
- (3) Part 23, effective February 1, 1965, Amendments 23-1 through 23.34, only applicable to Subpart C, excluding 23.571 and 23.572.
- (4) 14 CFR Part 23, effective February 1, 1965, only as applicable to turboprop engine installations and listed by Part 23 section below.  
(The Part 23 amendment level is shown in parentheses.)

23.49(e)(2)(-21)	23.933(-7)	23.1045(-7)	23.1305(-15)
23.65(c)(-21)	23.937(-7)	23.1091(-7)	23.1323(-7)
23.75(b)(-7)	23.951(-15)	23.1093(-15)	23.1337(-7)
23.77(b)(-21)	23.955(-7)	23.1103(-7)	23.1353(-20)
23.173(-14)	23.959(-7)	23.1105(0)	23.1521(0)
23.175(-14)	23.977(-17)	23.1111(-7)	23.1527(-7)
23.177(0)	23.991(-7)	23.1121(-7)	23.1529(-8)
23.371(-7)	23.997(-15)	23.1141(-14)	23.1545(-7)
23.629(e)(-31)	23.1013(-15)	23.1143(-7)	23.1549(-17)
23.831(0)	23.1015(-15)	23.1145(-18)	23.1557(-14)
23.901(-7)	23.1019(-15)	23.1155(-7)	23.1583(-10)
23.903(-14)	23.1027(-14)	23.1165(0)	23.1587(a)(-7)
23.905(0)	23.1041(-7)	23.1183(-15)	
23.929(-14)	23.1043(-7)	23.1303(0)	

Certification Basis (cont'd)	<p>(5) Exemption No. 4898 (CAR 3.83 - 70 mph stall speed) issued January 21, 1988. For S/N T65HG-013DC and up, compliance with FAA Policy Memorandum dated December 1, 1997, Section 23.49, has been shown (61 knot stall speed met with hopper empty), in lieu of previously required Exemption No. 4898.</p> <p>(6) Equivalent Safety Finding to §23.473(b), dated March 15, 1988 for 7,600 pound landing weight.</p> <p>(7) The intent of §25.305(c) regarding the dynamic response of the engine mount structure.</p>	
Airspeed Limits (CAS)	Vne (Never Exceed)	220 m.p.h. (191 knots)
	Vp (Maneuvering)	167 m.p.h. (145 knots)
	Vno (Max. Structural Cruising)	187 m.p.h. (163 knots)
	Vfe (Flap Extended)	157 m.p.h. (137 knots)
C.G. Range	<p>Forward Limit 7,600 pounds and below is +24.0 inches aft of datum          Forward limit at 10,500 pounds is 26 inches aft of datum with straight line variation to 7,600 pounds at 24.0 inches.          For S/N T65-002DC thru T65-012DC: Aft Limit at all weights is +29.0 inches aft of datum.          For S/N T65HG-013DC and subsequent: Aft limit at all weights is +28.0 inches aft of datum.          Datum is the leading edge of the wing.</p>	
Maximum Takeoff Weight	10,500 lbs.	
Maximum Landing Weight	7,600 lbs.	
Minimum Weight	5,000 lbs.	
Maximum Operating Altitude	12,000 feet	
Number of Seats	1 (+89) 1 (+127)	
Maximum Cargo Load	<p>See weight and balance data.          Maximum cargo compartment, 200 lbs. (+120).          Maximum hopper load, 4,000 lbs. (+29.9).</p>	
Fuel Capacity	228 gallon usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.	
Oil Tank Capacity	11 quarts - usable oil tank capacity 6 quarts.	
Control Surface Movements	Elevator	Up $27^{\circ} \pm 1^{\circ}$ Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$ Down $22^{\circ} \pm 1^{\circ}$
	Rudder	Left $24^{\circ} \pm 1^{\circ}$ Right $24^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$ Down $17^{\circ} \pm 1^{\circ}$
	Flaps	Down $15^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	<p>T65-002DC thru T65-012DC, and T65HG-013DC and subsequent.          For T65HG-011 and subsequent one-place configuration information see NOTE 25.</p>	

Required Equipment The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include for S/N T65-002DC thru T65-009DC Ayres Corporation Airplane Flight Manual approved June 8, 1988, and Supplement for Restricted Category Operation approved June 8, 1988, or for S/N T65HG-013DC and subsequent, Thrush Airplane Flight Manual approved October 28, 2005, or later approved versions.

Agricultural Dispersal Equipment Equipment eligible on S/N T65-002DC thru T65-012DC:  
High Volume Dispersal System, Ayres Dwg. No. 21563.  
See NOTE 17 for additional optional equipment.  
CAUTION: For operation with the Micronair Spray System or the Fire Bomber System, or with any system when an Agavenco pump is installed, the placards for airspeed limitations referred to in NOTE 2(q), 2(o), or 2(p), respectively, for the S2R are applicable.  
Equipment eligible on S/N T65HG-013DC and up:  
Standard Spray System, Thrush Dwg. No. 81071.

**XI-Model S2R-T45, 2 PCLM (Restricted Category Only)\*, Approved July 23, 1990**

\*See Note under certification basis for approved special purpose operations. See Note 15 for one-place configuration. See Note 22 for life limited parts.

Engine Pratt & Whitney (United Aircraft of) Canada PT6A-45, -45A, -45B, -45R (Dry ratings only)

Fuel See Airplane Flight Manual

Oil UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.

Engine Limits

	Takeoff and <u>Max. Cont.</u>	Transient (2 sec.) <u>Start/Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	1,173 (TO) 1,020 (MC)		900	
Torque (PSI)	43.3	61 Accel.		
ITT (°C)	800*	1,000 Start 850 Accel.	800	700
Ng (%)	104	104		52
Np (RPM)	1,700	1,870	1650	
Oil Press (PSIG)	100 to 135		100 to 135	60 min.
Oil Temp (°C)	10 to 99	-40 min. start 0 to 104 accel.	0 to 99	-40 to 99

\*760 °C for the PT6A-45 engine

Propeller and Propeller Limits (See Note 24 for pitch limits) Hartzell HC-B5MP-3C propeller, constant speed, feathering and reversing; Hub Model HC-B5MP-3C; Blade Model M-10876ANS.  
Diameter 111 inches maximum, 106.0 inches minimum.

Airspeed Limits (CAS)

Vne (Never Exceed)	159 mph (138 knots)
Vp (Maneuvering)	126 mph (109 knots)
Vno (Max. Structural Cruising)	126 mph (109 knots)
Vfe (Flap Extended)	123 mph (107 knots)
Maximum Dump Speed	120 mph (104 knots)

C. G. Range (+22.5) to (+27.5) without Elevator Down Spring, P/N 19661-1  
(+22.5) to (+29.0) with P/N 19661-1 installed.

Maximum Weight 6,000 lbs.

Maximum Operating Altitude	12,000 feet		
Number of Seats	1 (+89) (See NOTE 15 for one-place configuration.) 1 (+127)		
Maximum Cargo Load	Passenger/cargo compartment 200 lbs. maximum. See NOTE 15 for one-place configuration. Maximum hopper load, 4,000 lbs. (+29.9).		
Fuel Capacity	228 gallons usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.		
Oil Tank Capacity	11 quarts - usable oil tank capacity 6 quarts.		
Control Surface Movements (See Note 15 for one-place configuration)	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$
	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	T45-001DC and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved July 20, 1990, or later approved version.		
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems.		

**XII-Model S2R-G6, 1 PCLM (Restricted Category Only)\*, Approved March 5, 1992**

\*See Note under certification basis for approved special purpose operations. See Note 8 for two-place configuration. See Note 22 for life limited parts.

Engine	Honeywell (Garrett) TPE331-6				
Fuel	See Airplane Flight Manual				
Oil	MIL-L-23699B				
Engine Limits		Takeoff (5 min.)	Max. Continuous	Ground Idle	Starting
	SHP	750	715		
	Torque (%)	100	95		
	ITT ( $^{\circ}$ C)	923	923		1,149 max.
	RPM (%)*	100	100	65 to 85	
	Oil Press (PSIG)	70 to 120	70 to 120	40 to 120	
	Oil Temp ( $^{\circ}$ C)	55 to 127	55 to 127	-40 to 127	-40 to 127
		* Avoid operation between 18 and 28 percent RPM, except for transient during start and shutdown.			
Propeller and Propeller Limits (See Note 24 for pitch limits)	Hartzell propeller, Hub Model HCB3TN-5M, Blade Model T10282N+4. Diameter 106.0 inches maximum, 102.0 inches minimum.				
Airspeed Limits (CAS)	Vne (Never Exceed)	159 mph (138 knots)			
	Vp (Maneuvering)	126 mph (109 knots)			
	Vno (Max. Structural Cruising)	126 mph (109 knots)			
	Vfe (Flap Extended)	123 mph (107 knots)			
	Maximum Dump Speed	120 mph (104 knots)			

C. G. Range	(+26.5) to (+30.0) at 6,000 lbs. (+24.0) to (+30.0) at 4,000 lbs. Straight line variation in the forward limit between 4,000 lbs. and 6,000 lbs.		
Maximum Weight	6,000 lbs.		
Maximum Operating Altitude	12,000 feet		
Number of Seats	1 (+89) (See Note 8 for two-place configuration)		
Maximum Cargo Load	Maximum baggage compartment 60 lbs. See NOTE 8 for two-place configuration. Maximum hopper load, 4,000 lbs. (+29.9).		
Fuel Capacity	228 gallons usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.		
Oil Tank Capacity	8 quarts - usable oil tank capacity 7 quarts.		
Control Surface Movements	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$
	Rudder	Left $19^{\circ} \pm 1^{\circ}$	Right $19^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	G6-101 and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved March 5, 1992, or later approved version.		
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems.		

**XIII-Model S2R-G10, 1 PCLM (Restricted Category Only)\*, Approved January 12, 1993**

\*See Note under certification basis for approved special purpose operations. See Note 8 for two-place configuration. See Note 22 for life limited parts.

Engine	Honeywell (Garrett) TPE331-10
Fuel	See Airplane Flight Manual
Oil	MIL-L-23699B
Engine Limits	

	Takeoff (5 min.)	Max. Continuous	Ground Idle	Starting
SHP	900	900		
Torque (%)	100	100		
EGT ( $^{\circ}$ C)	**	**	**	770 max.
RPM (%) *	100	100	72 to 85	
Oil Press (PSIG)	70 to 120	70 to 120	40 to 120	
Oil Temp ( $^{\circ}$ C)	55 to 127	55 to 110	-40 to 110	-40 to 110

\*Avoid operation between 18 and 28 percent RPM, except for transient during start and shutdown.

\*\*EGT Limits: 600 $^{\circ}$ C EGT at 45 $^{\circ}$ C OAT and 540 $^{\circ}$ C EGT at -15 $^{\circ}$ C OAT, straight line variation in between.

Propeller and Propeller Limits (See Note 24 for pitch limits)	McCauley Hub Model 4HFR34C653-[X], Blade Model [X]-L106FA-0. Diameter 106.0 inches maximum, 105.0 inches minimum.		
	McCauley Hub Model 4HFR34C662-[X], Blade Model [X]-L108FA-0. Diameter 108.0 inches maximum, 105.0 inches minimum.		
	Hartzell Model HC-B4TN-5NL, Blade Model LT10890N. Diameter 109.5 inches maximum, 107.5 inches minimum.		
Airspeed Limits (CAS)	Vne (Never Exceed)	159 mph (138 knots)	
	Vp (Maneuvering)	126 mph (109 knots)	
	Vno (Max. Structural Cruising)	126 mph (109 knots)	
	Vfe (Flap Extended)	123 mph (107 knots)	
	Maximum Dump Speed	120 mph (104 knots)	
C. G. Range	(+26.5) to (+30.0) at 6,000 lbs. (+24.0) to (+30.0) at 4,000 lbs. Straight line variation in the forward limit between 4,000 lbs. and 6,000 lbs.		
Maximum Weight	6,000 lbs.		
Maximum Operating Altitude	12,000 feet		
Number of Seats	1 (+89) (See NOTE 8 for two-place configuration)		
Maximum Cargo Load	Maximum baggage compartment 60 lbs. See NOTE 8 for two-place configuration. Maximum hopper load, 4,000 lbs. (+29.9).		
Fuel Capacity	228 gallons usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.		
Oil Tank Capacity	8 quarts - usable oil tank capacity 7 quarts.		
Control Surface Movements	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$
	Rudder	Left $19^{\circ} \pm 1^{\circ}$	Right $19^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Flaps		Down $15^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	G10-101 and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved January 12, 1993, or later approved version.		
Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems.		
<b><u>XIV-Model S2R-G5, 1 PCLM (Restricted Category Only)*, Approved August 20, 1993</u></b>			
*See Note under certification basis for approved special purpose operations. Also see Note 8 for two-place configuration. See Note 22 for life limited parts.			
Engine	Honeywell (Garrett) TPE331-5		
Fuel	See Airplane Flight Manual		
Oil	MIL-L-23699B		

## Engine Limits

	<u>Takeoff (5 min.)</u>	<u>Max. Continuous</u>	<u>Ground Idle</u>	<u>Starting</u>
SHP	750	715		
Torque (%)	100	95		
ITT (°C)	923	923		1,149 max.
RPM (%)*	100	100	72 to 85	
Oil Press (PSIG)	70 to 120	70 to 120	40 to 120	
Oil Temp (°C)	55 to 127	55 to 110	-40 to 110	-40 to 110

\* Avoid operation between 18 and 28 percent RPM, except for transient during start and shutdown.

Propeller and Propeller Limits  
(See Note 24 for pitch limits)

McCauley Hub Model 4HFR34C653-[X], Blade Model [X]-L106FA-0.  
Diameter 106.0 inches maximum, 105.0 inches minimum.

McCauley Hub Model 4HFR34C662-[X], Blade Model [X]-L108FA-0.  
Diameter 108.0 inches maximum, 105.0 inches minimum.

Hartzell Model HC-B4TN-5NL, Blade Model LT10890N.  
Diameter 109.5 inches maximum, 107.5 inches minimum.

## Airspeed Limits (CAS)

Vne (Never Exceed)	159 mph (138 knots)
Vp (Maneuvering)	126 mph (109 knots)
Vno (Max. Structural Cruising)	126 mph (109 knots)
Vfe (Flap Extended)	123 mph (107 knots)
Maximum Dump Speed	120 mph (104 knots)

## C. G. Range

(+26.5) to (+30.0) at 6,000 lbs.  
(+24.0) to (+30.0) at 4,000 lbs.  
Straight line variation in the forward limit between 4,000 lbs. and 6,000 lbs.

## Maximum Weight

6,000 lbs.

## Maximum Operating Altitude

12,000 feet

## Number of Seats

1 (+89) (See NOTE 8 for two-place configuration)

## Maximum Cargo Load

Maximum baggage compartment 60 lbs. See NOTE 8 for two-place configuration.  
Maximum hopper load, 4,000 lbs. (+29.9).

## Fuel Capacity

228 gallons usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.

## Oil Tank Capacity

8 quarts - usable oil tank capacity 7 quarts.

## Control Surface Movements

Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$
Rudder	Left $19^{\circ} \pm 1^{\circ}$	Right $19^{\circ} \pm 1^{\circ}$
Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
Flaps		Down $15^{\circ} \pm 1^{\circ}$

## Serial Numbers Eligible

G5-101 and subsequent

## Required Equipment

The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved August 20, 1993, or later approved version.

## Agricultural Dispersal Equipment

See NOTE 17 for dispersal systems.

**XV-Model S2R-G1, 1PCLM (Restricted Category Only\*), Approved August 29, 1995.**

\*See Note under certification basis for approved special purpose operations. See Note 22 for life limited parts.

Engine	Honeywell (Garrett) TPE331-1				
Fuel	See Airplane Flight Manual				
Engine Limits		<u>Takeoff (5 min)</u>	<u>Max. Continuous</u>	<u>Ground Idle</u>	<u>Starting</u>
	SHP	665	665		
	Torque (%)	100	100		
	EGT (°C)	**	**	**	
	RPM (%)*	100	100	65 to 85	
	Oil Pressure (PSIG)	70 - 130	70 - 120	40 - 120	
	Oil Temp (°C)	55 - 127	55 - 110	-40 to 110	-40 to 110
		* Avoid operation between 18 and 28 percent RPM except for transient during start and shutdown.			
		** EGT Limits: 600°C EGT at 120°F OAT and 520°C EGT at 0°F OAT, straight line variation between.			
Propeller and Propeller Limits (See Note 24 for pitch limits)	Hartzell Hub Model HCB3TN-5M, Blade Model T10282N+4. Diameter 106.0 inches maximum, 102.0 inches minimum.				
Airspeed Limits (CAS)	$V_{NE}$ (Never Exceed)	159 mph (138 knots)			
	$V_a$ (Maneuvering)	126 mph (109 knots)			
	$V_{no}$ (Max. Structural Cruising)	126 mph (109 knots)			
	$V_{fe}$ (Flap Extended)	123 mph (107 knots)			
	Maximum Dump Speed	120 mph (104 knots)			
C.G. Range	(+26.5) to (+30.0) at 6,000 lbs. (+24.0) to (+30.0) at 4,000 lbs. Straight line variation in the forward limit between 4,000 lbs. and 6,000 lbs.				
Maximum Weight	6,000 lbs.				
Maximum Operating Altitude	12,000 feet				
Number of Seats	1 (+89)				
Maximum Cargo Load	See weight and balance data. Maximum baggage compartment, 60.0 lbs. Maximum hopper load, 4,000 lbs. (+29.9).				
Fuel Capacity	104 gallons usable (one 53 gallon tank in each wing, tanks interconnected). 140 gallons usable (one 69 gallon tank in each wing, tanks interconnected). See NOTE 1 for data on unusable fuel.				
Oil Tank Capacity	8 quarts - usable oil tank capacity 7 quarts.				
Control Surface Movements	Elevator	Up $27^\circ \pm 1^\circ$	Down $17^\circ \pm 1^\circ$		
	Elevator Tab	Up $8^\circ \pm 1^\circ$	Down $22^\circ \pm 1^\circ$		
	Rudder	Left $19^\circ \pm 1^\circ$	Right $19^\circ \pm 1^\circ$		
	Aileron	Up $21^\circ \pm 1^\circ$	Down $17^\circ \pm 1^\circ$		
	Flaps		Down $17^\circ \pm 1^\circ$		
Serial Numbers Eligible	G1-101 and subsequent.				

Required Equipment The basic required equipment as prescribed in the applicable airworthiness regulation (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved August 29, 1995, or later approved revision.

Agricultural Dispersal Equipment See NOTE 17 for dispersal systems.

**XVI-Model S2RHG-T34, 2 PCLM (Restricted Category Only)\*, Approved November 5, 1997**

\*See Note under Certification Basis for approved special purpose operations. See Note 15 for one-place configuration. See Note 22 for life limited parts.

Engine Pratt & Whitney (United Aircraft of) Canada PT6A-34AG  
Optional Engines: Pratt & Whitney Canada PT6A-34 (See NOTE 12 for information)  
Pratt & Whitney Canada PT6A-36 (Dry Configuration Only)  
Pratt & Whitney Canada PT6A-41, PT6A-41AG, and PT6A-42 (See NOTE 14 for information)

Fuel Jet A, Jet B, JP-4, JP-5, Automotive Diesel Number 1D or 2D in accordance with UACL Service Bulletin Number 1344. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.) Automotive diesel fuel is approved only for agricultural application flights and only when the free air temperature is above:  
+20°F for Grade No. 1D  
+40°F for Grade No. 2D

Oil UACL PT6 Engine Service Bulletin Number 1001 lists approved brands of oil.

Engine Limits PT6A-34AG/-34/-36:

	Takeoff and <u>Max. Cont.</u>	Transient <u>Start/Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	750			
Torque(PSI)(2sec)	64.5	68.4 Trans	64.5	
ITT (°C)	790	1,090 Start (2 sec.)	790	
Ng (%)	101.6	102.6 Trans (2 sec.)	101.6	
Np (RPM)	2,200	2,420 Trans (2 sec.)	2,100	
Oil Press(PSIG)	85 to 105	85 to 100	85 to 105	40 min.
Oil Temp (°C)	10 to 99	-40 min.	0 to 99	-40 to 99

The ratings shown are based on the static sea level standard condition with no external accessory loads and no air bleed.

Propeller and Propeller Limits (See Note 24 for pitch limits) Hartzell Hub Model HC-B3TN-3C (or HC-B3TN-3D) with Blade Model T-10282, Diameter 102.5 inches maximum, 92.5 inches minimum or optional Blade Model T-10282N+4, Diameter 106 inches maximum, 98 inches minimum.

Certification Basis

- (1) 14 CFR Part 21.25(a)(1)
- (2) CAR 3, effective May 15, 1956, including Amendments 3-1 through 3-8 as modified by CAR 8.10(a)(1) effective October 11, 1950, except the following paragraphs (allowed under § 21.25(a)(1)):  
CAR 3.83  
CAR 3.780(a)(3)  
CAR 3.780(a)(4)
- (3) Part 23, effective February 1, 1965, Amendments 23-1 through 23-34, only applicable to Subpart C, and other Part 23 sections listed below.

- (4) Part 23, effective February 1, 1965, only as applicable to turboprop engine installations and listed by Part 23 section below.  
(The Part 23 amendment level is shown in parentheses.)

23.49(e)(2)(-21)	23.907(0)	23.1041(-7)	23.1183(-15)
23.65(c)(-21)	23.929(-14)	23.1043(-7)	23.1303(0)
23.75(b)(-7)	23.933(-7)	23.1045(-7)	23.1305(-15)
23.77(b)(-21)	23.937(-7)	23.1091(-7)	23.1323(-7)
23.173(-14)	23.951(-15)	23.1093(-15)	23.1337(-7)
23.175(-14)	23.955(-7)	23.1103(-7)	23.1353(-20)
23.177(0)	23.959(-7)	23.1105(0)	23.1521(0)
23.371(-7)	23.977(-17)	23.1111(-7)	23.1527(-7)
23.572(a)(1)(-34)	23.991(-7)	23.1121(-7)	23.1529(-8)
23.629(e)(-31)	23.997(-15)	23.1141(-14)	23.1545(-7)
23.831(0)	23.1013(-15)	23.1143(-7)	23.1549(-17)
23.901(-7)	23.1015(-15)	23.1145(-18)	23.1557(-14)
23.903(-14)	23.1019(-15)	23.1155(-7)	23.1583(-10)
23.905(0)	23.1027(-14)	23.1165(0)	23.1587(a)(-7)

- (6) Equivalent Safety Finding to §23.473(b), dated March 15, 1988 for 7,650 pound landing weight and Equivalent Level of Safety Finding No. ACE-04-05 dated July 26, 2004 for 8,800 pound landing weight.

- (7) The intent of §25.305(c) regarding the dynamic response of the engine mount structure.

Airspeed Limits (CAS)	Vne (Never Exceed)	190 mph (165 knots)
	Vp (Maneuvering)	154 mph (134 knots)
	Vno (Max. Structural Cruising)	162 mph (141 knots)
	Vfe (Flap Extended)	144 mph (125 knots)
C. G. Range	Forward limit at 7,600 lbs. and below, +22.5 inches aft of datum. Forward limit at 9,500 lbs., +26.0 inches aft of datum. (Straight line variation in the forward limit between 7,600 and 9,500 lbs.) Aft limit +29.0 inches aft of datum. Datum is the leading edge of the wing.	
Maximum Weight	9,500 lbs.	
Maximum Landing Weight	7,650 lbs. for S/N T34HG-101 and T34HG-102 aircraft equipped with Main Landing Gear p/n 50111-503/-504/-505/-506 8,800 lbs. for S/N T34HG-103 and up equipped with Main Landing Gear p/n 94200	
Minimum Weight	5,000 lbs.	
Maximum Operating Altitude	12,000 feet	
Number of Seats	1 (+89) (See NOTE 15 for one-place configuration.) 1 (+127 forward facing) or (+111 aft facing)	
Maximum Cargo Load	See weight and balance data. Maximum passenger/cargo compartment, 200 lbs. (+120). (See NOTE 15 for one-place configuration) Maximum hopper load, 4,000 lbs. (+29.9).	
Fuel Capacity	228 gallons usable, one 115 gallon tank in each wing, tanks interconnected. See NOTE 1 for data on unusable fuel.	
Oil Tank Capacity	11 quarts - usable oil tank capacity 6 quarts.	

Control Surface Movements	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$
	Rudder	Left $24^{\circ} \pm 1^{\circ}$	Right $24^{\circ} \pm 1^{\circ}$
	S/N T34HG-103 and up:		
		Left $22^{\circ} \pm 1^{\circ}$	Right $22^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
Flaps		Down $15^{\circ} \pm 1^{\circ}$	
Serial Numbers Eligible	T34HG-101DC and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved November 5, 1997, or later approved versions. For S/N T34HG-103 and up, Thrush Aircraft, Inc. Airplane Flight Manual dated April 16, 2004, or later FAA approved revision, is required.		

Agricultural Dispersal Equipment	See NOTE 17 for dispersal systems. CAUTION: For operation with the Micronair Spray Equipment System or the Fire Bomber System, or with any system when an Agavenco pump is installed, the placards for airspeed limitations referred to in NOTE 2(q), 2(o), or 2(p), respectively, for the S2R are applicable.
----------------------------------	---

**XVII-Model S2R-T660, 1 PCLM (Restricted Category Only)\*, Approved March 13, 2000**

\*See Note under Certification Basis for approved special purpose operations. See Note 8 for two-place configuration.

Engine	Pratt & Whitney Canada PT6A-60AG Optional Engines: Pratt & Whitney Canada PT6A-65AG, -65AR, -65B (-65AR must have automatic power reserve feature disabled) Pratt & Whitney Canada PT6A-45A, -45B, -45R eligible on S/N T660-108 and up Pratt & Whitney Canada PT6A-67AG eligible on S/N T660-109 and up
Fuel	Jet A, Jet B, JP-4, JP-5, Automotive Diesel Number 1D or 2D in accordance with P&WC Specifications CPW 204, CPW 46, CPW 381, and Service Bulletin 13244. (If jet fuel is not available, aviation gasoline, MIL-G-5572, all grades, may be used for a maximum of 150 hours between overhauls.) Automotive diesel fuel is approved only for agricultural application flights and only when the free air temperature is above: +20°F for Grade No. 1D +40°F for Grade No. 2D
Oil	UACL PT6 Engine Service Bulletin Number 1001, 3001, 4001, 11001, 12002 and 13001 lists approved brands of engine oil.

Engine Limits

<u>PT6A-60AG:</u>	Transient*					
	<u>Takeoff</u>	<u>Max. Cont.</u>	<u>Start</u>	<u>Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	1,050	1,020			900	
Torque (PSI)**	38.8	37.7		61		
ITT (°C)	820	775	1,000	850	760	750
Ng (%)	104	104	104	104		58
Np (RPM)	1,700	1,700		1,870	1,650	
Oil Press (PSIG)	90 to 135	90 to 135	0 to 200	40 to 200	90 to 135	60 Minimum
Oil Temp (°C)	0 to 110	0 to 110	-40 Min.	0 to 110	0 to 99	-40 to 110

\*Transient engine limits are 5 seconds for starting and 20 seconds for acceleration.

\*\*The Torque pressure limits listed are for NP=1,700 RPM only.

## Engine Limits (cont'd)

PT6A-45A/-45B/-45R:

	<u>Takeoff</u>	<u>Max.Cont.</u>	<u>Start</u>	<u>Transient*</u>		<u>Reverse</u>	<u>Idle</u>
				<u>Accel.</u>			
SHP	1,050	1,020				900	
Torque (PSI)**	38.8	37.7					
ITT (°C)	800	765	1,000	850	760		750
Ng (%)	104	104		104			56
Np (RPM)	1,700	1,700		1,870	1,650		
Oil Press (PSIG)	90 to 135	90 to 135	0 to 200	40 to 200	90 to 135		60 min.
Oil Temp (°C)	10 to 99	10 to 99	-40 min.	10 to 99	0 to 99		-40 to 99

PT6A-67AG:

	<u>Takeoff</u>	<u>Max.Cont.</u>	<u>Start</u>	<u>Transient*</u>		<u>Reverse</u>	<u>Idle</u>
				<u>Accel.</u>			
SHP	1,300	1220				900	
Torque (PSI)**	48.0	45.1		61			
ITT (°C)	800	800	1,000	850	760		750
Ng (%)	104	104	104	104			51
Np (RPM)	1,700	1,700		1,870	1,650		
Oil Press (PSIG)	90 to 135	90 to 135	0 to 200	40 to 200	90 to 135		60 min.
Oil Temp (°C)	0 to 110	0 to 110	-40 min.	0 to 110	10 to 105		-40 to 110

\*Transient engine limits are 5 seconds for starting and 20 seconds for acceleration.

\*\*The Torque pressure limits listed are for NP=1,700 RPM only.

PT6A-65AG/-65AR/-65B:

	<u>Takeoff</u>	<u>Max.Cont.</u>	<u>Start</u>	<u>Transient*</u>		<u>Reverse</u>	<u>Idle</u>
				<u>Accel.</u>			
SHP	1,300 <sup>†</sup>	1,220 <sup>†</sup>				900	
Torque (PSI)**	48.0	45.1		61*			
ITT (°C)	810	810	1,000*	850*	760		750 <sup>††</sup>
Ng (%)	104	104		104			56
Np (RPM)	1,700	1,700		1,870	1,650		
Oil Press (PSIG)	90 to 135	90 to 135	0 to 200	40 to 200	90 to 135		60 min.
Oil Temp (°C)	0 to 110 <sup>†††</sup>	0 to 110 <sup>†††</sup>	-40 min.	-40 to 110	0 to 110 <sup>†</sup>		-40 to 110

\*Transient engine limits are 5 seconds for starting and 20 seconds for acceleration.

\*\*The Torque pressure limits listed are for NP=1,700 RPM only.

<sup>†</sup> For PT6A-65B 1100 SHP Take off & Max. Continuous, Torque 43.34 psi Takeoff & max continuous, ITT 820°C Take off; 0 to 99 (°C) Oil Temp Maximum Reverse. For PT6A-65AR ITT 810°C Take off, 810°C Max continuous, 10 to 105 (°C) Oil Temp Maximum Reverse.

<sup>††</sup> For PT6A-65B 700°C, For PT6A-65AR 715°C.

<sup>†††</sup> For PT6-65AR 10 to 110.

## Propeller and Propeller Limits

For PT6A-60AG/-45/-45A/-45B/-45R:

Hartzell HC-B5MP-3C propeller, constant speed, feathering and reversing;  
Hub Model HC-B5MP-3C with Blade Model M10876ANS or M10876AS  
Diameter 111.2 inches maximum, 110.7 inches minimum.  
Pitch (42 in. Sta.) 16.5° low, 79.0° feather, -11.0° reverse

## Propeller and Propeller Limits (cont'd)

For PT6A-67AG:

Hartzell HC-B5MA-3D propeller, constant speed, feathering and reversing;  
 Hub Model HC-B5MA-3D with Blade Model M11276NS  
 Diameter 115.2 inches maximum, 114.7 inches minimum.  
 Pitch (42 in. Sta.) 13.9° low, 83.1° feather, -10.0° reverse

For PT6A-65AG/-65AR/-65B:

Hartzell HC-B5MP-3F propeller, constant speed, feathering and reversing;  
 Hub Model HC-B5MP-3F with Blade Model M11276NS  
 Diameter 115.2 inches maximum, 114.7 inches minimum.  
 Pitch (42 in. Sta.) 13.9° low, 83.1° feather, -10.0° reverse

## Certification Basis

- (1) 14CFR Part 21.25(a)(1)
- (2) 14CFR Part 23 -  
 Subpart A, Amendment 23-53;  
 Subpart B, Amendment 23-53;  
 Subpart C, Amendment 23-53 with Appendix A, but §§23.423, 23.425, 23.427, 23.441, 23.443, and 23.455 are at Amendment 23-34;  
 Subpart D, Amendment 23-53, but §23.607 is at Amendment 23-34, §23.629 is at Amendment 23-31, and §§23.785, 23.787, 23.807, 23.853, 23.863, 23.865 and 23.867 are at Amendment 23-14;  
 Subpart E, Amendment 23-14;  
 Subpart F, Amendment 23-0;  
 Subpart G, Amendment 23-53;  
 except those regulations found inappropriate for restricted category agricultural airplanes as listed in FAA Advisory Circular 21.25-1, dated December 1, 1997, and compliance with regulations listed in ACE-110 policy memorandum, dated December 1, 1997, demonstrated in accordance with that memorandum.

## Airspeed Limits (CAS)

V <sub>ne</sub> (Never Exceed)	219 mph (191 knots)
V <sub>A</sub> (Maneuvering)	160 mph (140 knots)
V <sub>no</sub> (Max. Structural Cruising)	206 mph (180 knots)
V <sub>fe</sub> (Flap Extended)	143 mph (126 knots)

(See NOTE 23 and Airplane Flight Manual for decreased airspeeds at 14,150 lbs.)

## C. G. Range

Forward Limit at 12,500 lbs. is 24 inches aft of datum with straight line variation to 8,000 lbs. at 27 inches aft of datum.  
 Forward Limit below 8,000 pounds is 27 inches aft of datum.  
 Aft Limit at 12,500 lbs. is 27 inches aft of datum with straight line variation to 8,000 lbs. at 30 inches aft of datum.  
 Aft Limit below 8,000 lbs. is +30.0 inches aft of datum.  
 Datum is the leading edge of the wing.

## Maximum Takeoff Weight

12,500 lbs. (See NOTE 23 for limitations at 14,150 lbs.)

## Maximum Landing Weight

12,500 lbs.

## Minimum Weight

6,100 lbs.

## Maximum Operating Altitude

12,000 feet

## Number of Seats

1 (+89) (See NOTE 8 for two-place configuration)

## Maximum Cargo Load

See weight and balance data.  
 Maximum baggage compartment load is 200 lbs.(+112).  
 Maximum hopper load, 5,500 lbs.(+20.6).

## Fuel Capacity

225.6 gallons usable, one 115 gallon tank in each wing, tanks interconnected.  
 See NOTE 1 for data on unusable fuel.

Oil Tank Capacity	10 U.S. quarts - usable oil tank capacity 6 quarts.		
Control Surface Movements	Elevator	Up $27^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Elevator Tab	Up $8^{\circ} \pm 1^{\circ}$	Down $22^{\circ} \pm 1^{\circ}$
	Rudder	Left $19^{\circ} \pm 1^{\circ}$	Right $19^{\circ} \pm 1^{\circ}$
		S/N T660-109 and up:	
		Left $22^{\circ} \pm 1^{\circ}$	Right $22^{\circ} \pm 1^{\circ}$
	Aileron	Up $21^{\circ} \pm 1^{\circ}$	Down $17^{\circ} \pm 1^{\circ}$
	Flaps	Down $15^{\circ} \pm 1^{\circ}$	
Serial Numbers Eligible	T660-101 and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Ayres Corporation Airplane Flight Manual approved March 13, 2000, or later approved revision. For S/N T660-109 and up, Thrush Aircraft, Inc. Airplane Flight Manual dated March 5, 2004, or later FAA approved revision, is required. (See NOTE 23 for operation at 14,150 lbs.)		
Agricultural Dispersal Equipment	Standard Spray System, Ayres Dwg. No. 95340 Spreader Installation, Ayres Dwg. No. 95370 Transland Hydraulic Fire Door Installation, Ayres Dwg. No. 95385		
Structural Limitations	The following parts must be replaced at the times in service indicated: (See NOTE 23 for life limits if ever operated between 12,500 lbs. and 14,150 lbs.)		
	<u>Part Name</u>	<u>Part Number</u>	<u>Life Limit</u>
	Rear Spar Doubler, Lower	95627-3	20,000
	Rear Spar, Inboard, L&R	95623-1/-2	20,000
	Aft Main Spar Lug, L&R	95605-1/-2	21,750
	Forward Main Spar Lug, L&R	95606-1/-2	20,000
	Spar Cap Assy, L&R	95603-1/-2	26,625
	Steel Doubler Plate	95614-1	38,400
<u>Data Pertinent to All Models</u>			
Certification Basis	CAR 8 effective October 11, 1950, restricted category.		
For All Models	Type certificate A4SW issued November 1, 1965.		
Except for S2R-T65, S2RHG-T65, S2RHG-T34 and S2R-T660 see Sections IX, X, XVI and XVII	For All Models approved after January 1, 1980: Model S2R-R1340 and other models certificated after January 1, 1980, are only to be certificated for the special purposes of agricultural aircraft operations as defined in 14 CFR Part 137.3 and aerial dispensing of liquids for fires due to 14 CFR Part 36.1(a)(2) requirements. Aircraft noise measurement is not required for these special purpose operations and aircraft noise has not been shown to comply with the noise requirements in 14 CFR Part 36.		
Production Basis	Production Certificate Number 5SO.		
Export Eligibility	Aircraft will be eligible for issuance of an Export Certificate of Airworthiness subject to compliance with Federal Aviation Regulations Part 21, Subpart L, Sections 21.321 through 21.339. Special requirements of specific foreign countries are contained in Advisory Circular 21-2D. Note: Restricted category aircraft may not be operated in a foreign country without the express written approval of that country. These aircraft have not been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 of the Convention on International Civil Aviation.		

Datum	Wing leading edge.
Leveling Means	Lower longeron below cockpit.
Number of Passengers	None

NOTE 1: Current weight and balance report including list of equipment included in certificated empty weight, and loading instructions when necessary, must be provided for each aircraft at the time of original certification. The empty weight and corresponding center of gravity location must include the following unusable fuel:

Model 600 S2D, all serial numbers	54 lbs. at (+38.5)
Model S2R, S/N 1380R	24 lbs. at (+38.5)
Model S2R, S/N 1416R and 1418R	36 lbs. at (+38.5)
Model S2R, S/N 1419R thru 1499R, 1501R thru 1510R	48 lbs. at (+38.5)
Model S2R, S/N 1500R, 1511R thru 4999R, 5000R and subsequent	18 lbs. at (+38.5)
Models S2R-T34, S2R-T15, S2R-T65, S2RHG-T65, S2R-R3S, S2R-T11, S2R-R1340, S2R-R1820, S2R-T45, S2R-G6, S2R-G10, S2R-G5, S2R-G1, and S2RHG-T34, all serial numbers	18 lbs. at (+38.5)
Model S2R-T660, all serial numbers	30 lbs. at (+38.5)

NOTE 2: The following information on placards pertaining to flight and operating instructions and limitations must be displayed in full view of the pilot:

- (a) "Restricted"
- (b) "This airplane must be operated as a restricted category airplane in accordance with the operating limitations stated in the form of placards and the Airplane Flight Manual."
- (c) "No acrobatic maneuvers including spins approved."
- (d) (1) Model 600 S2D: "The operation of this airplane is limited to day VFR conditions. Flight into known icing conditions prohibited."  
Model 600 S2D is eligible for day and night VFR conditions if approved light system, Snow Dwg. No. 90119 and 90132, is installed, in which case placard under NOTE 2(d)(2) applies.
- (2) Model S2R: "The operation of this airplane is limited to day and night VFR conditions. Flight into known icing conditions prohibited."
- (e) S2D and S2R only: "Design Maneuvering Speed: 126 mph  
Maximum Crosswind Velocity: 15 mph"  
"Maximum flap-down speed: 123 mph" (S2R only)
- (f) S2D and S2R only: "Avoid continuous ground operation between 1,280 and 1,900 R.P.M."
- (g) Adjacent to stall warning switch when dry battery stall warning system is installed (S/N 1311D thru 1415D, S/N 1380R, 1416R thru 1440R):  
"Stall warning switch must be on in flight. Change battery every four months to dated Eveready 6V No. 1461. Mark date battery changed on battery."
- (h) Adjacent to stall warning switch when 12 or 24 volt electrical system installed (S/N 1311D thru 1415D, S/N 1380R, 1416R thru 1440R):  
"Stall warning system is inoperative with generator and battery switches off."

- (i) When stall warning system is installed (S/N 1311D thru 1415D, S/N 1380R, 1416R thru 1440R):  
“Stall warning light -- test light daily before flight by moving lift indicator until light comes on.”
- (j) When canopy is installed: “No smoking”
- (k) Park brake: “On, depress pedals and pull lever. Off, depress pedals”
- (l) When locking tail wheel is installed: “Push stick forward to unlock tail wheel.”
- (m) Usable tank capacity (See "Fuel Capacity")
- (n) S2D Only - when Snow Spreader, Dwg. No. 80188, or Small Swathmaster, Dwg. No. 80187, is installed:  
“When the Snow Spreader or Small Swathmaster dispersal systems are installed, the following airspeed limitations must be observed:  
Maximum Maneuvering Speed 111 mph CAS  
Never Exceed Speed 140 mph”
- (o) S2R Only - when the Fire Bomber System, Dwg. No. 80792 for S/N 1416R thru 1576R or Dwg. No. 81069 for S/N 1577R and subsequent, is installed the following airspeed limitations must be observed:  
“With Fire Bomber Dump System installation and any disposal load, do not exceed 120 mph CAS.”
- (p) S2R Only (Agavenco Pump Only): “Do not operate pump above 115 mph CAS”
- (q) S2R Only: “Do not operate Micronair Units above 125 mph (CAS)”
- (r) The following placard must be displayed on the wings and adjacent to the fuel filler caps:  
“FUEL (\*) US GAL. MIN. OCTANE 87 FUEL TANKS ARE INTERCONNECTED - ALLOW SUFFICIENT TIME FOR FUEL LEVEL TO EQUALIZE BEFORE TOP-OFF OF TANKS. NO AROMATIC FUEL.”  
\*54.5 for 600 S2D Models, 35 for S/N 1380R Model S2R,  
53 for S/N 1416R and subsequent (See NOTE 9 for other approved fuel capacities)
- (s) The following placard must be displayed adjacent to the oil filler cap:  
“OIL TANK (\*) GAL. CAP.”  
\*9.2 for S2R, 10.9 for 600-S2D
- (t) “Sulphur dusting is prohibited unless special fire prevention measures have been incorporated in the aircraft.”
- (u) Placards for Model S2R-R1820 -
1. In clear view of the pilot:  
"ENGINE OPERATION LIMITS  
Takeoff only (1 min.) 2,500 RPM at 45.5" hg. at sea level (1,200 H.P.)  
For all other Operations - 1,000 H.P. at 2,300 RPM, 39.5 in. hg. & S.L.  
- 1,000 H.P. at 2,300 RPM, 37.2 in. hg. at 6,900 ft.  
STRAIGHT LINE VARIATION BETWEEN POINTS GIVEN  
100/130 MIN. GRADE AVIATION GASOLINE".
  2. At aux. fuel pump circuit breaker switch:  
"AUX. FUEL PUMP ON  
OFF"
  3. On throttle quadrant:  
- MAN RICH  
- AUTO RICH  
"MIXTURE - AUTO LEAN  
- FUEL CUTOFF".

4. At the primer switch and at the appropriate detent:  
"PRIMER ON  
OFF"
5. At the auxiliary fuel pump switch:  
"AUX. PUMP ON  
OFF"
6. At circuit breaker:  
"PRIMER  
2 AMPS"
7. At generator circuit breaker (if newly installed):  
"CB GEN." (50 AMP)."
8. At fuel filler caps:  
"Fuel 96 U.S. Gal. Min. Octane 100/130. Aviation Gasoline Fuel Tanks are interconnected. Allow sufficient time for fuel level to equalize before top-off of tank. No Aromatic Fuel." (See NOTE 9 for other approved fuel capacities)
9. On inside of oil tank filler door:  
"Oil Grade - Aero Shell or equivalent Above 32°F 120: Below 32°F 100  
NOTE: Detergent oil W120 and W100 may be used after a 50 hour break-in period on new piston rings. New rings must be seated on non detergent oil. Capacity 13 gallons."  
(See NOTE 15 for one-place configuration oil capacity.)
10. If not already installed on instrument panel, at Stall Warning Light and fuse (1 AMP):  
"STALL WARNING"
11. With H.S. 43D50/6933A-9 propeller only. Adjacent to the tachometer:  
"AVOID CONTINUOUS GROUND OPERATION BETWEEN 1200 AND 1500 RPM  
AND BETWEEN 1,900 AND 2,200 RPM"
12. For dual cockpit aircraft, in rear passenger/cargo area (all models except S2R-T65 and S2RHG-T65):  
"PASSENGER OR CARGO 200 LBS. MAXIMUM"
13. For dual cockpit S2R-T65 and S2RHG-T65, in rear passenger/cargo area:  
"CARGO 200 LBS. MAXIMUM"

Also see the FAA approved Airplane Flight Manual for required placards. (Not applicable to early Model 600 S2D and S2R which were not equipped with Airplane Flight Manuals.)

NOTE 3: Maintenance should be done in accordance with the maintenance manual for each model. Later models have mandatory replacement information and mandatory inspections in the Airworthiness Limitations sections of the manuals.

NOTE 4: Refer to Type Certificate Data Sheet Number A3SW for conditions and limitations applicable to the "Normal Category", Models 600 S2D, S-2R, S2R-T34, S2R-T15, S2R-T11, S2R-R3S, and S2R-R1340. The configuration of airplanes eligible for a standard airworthiness certificate is controlled by Thrush Drawing 30007.

NOTE 5. Model S2R, Optional Engine Installation  
(Only sections different from II are shown.)

Engine Wright R-1300-1B

Fuel 100/130 Minimum grade aviation gasoline

## Engine Limits

	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>
Takeoff (1 min.)	800	2,600	44.0	S.L.
Takeoff (1 min.)	800	2,600	42.5	3,500
Max. Continuous	700	2,400	39.5	S.L.
Max. Continuous	700	2,400	38.0	5,000

## Propeller and Propeller Limits

Hamilton Standard, constant speed, 3D40 Hub, (as modified by STC SP148NW)  
EAC-AG100-0S blades.

Diameter 108 5/16 inches maximum, 106 5/16 inches minimum.

Pitch settings, 23° low and 38.0° high at 42 inch station.

Governor, Hamilton Standard 4M-12-5

or

Hamilton Standard, constant speed, 23D40 Hub, 6601A-30S blades.

Diameter 108 inches maximum, 106 inches minimum.

Pitch settings, 24.5° low and 44.5° high at 42 inch station.

Governor, Hamilton Standard 4G-10-5

## C.G Range

(+22.5) to (+28.0)

## Control Surface Movements

Elevator	Up 27° ± 1°	Down 17° ± 1°
Elevator Tab	Up 8° ± 1°	Down 22° ± 1°
Rudder	Left 24° ± 1°	Right 24° ± 1°
Aileron	Up 21° ± 1°	Down 17° ± 1°
Flaps		Down 26° - 30°

## Serial Numbers Eligible

5000R and subsequent

## Certification Basis

CAR 8 effective October 11, 1950, restricted category. Type Certificate A4SW issued November 1, 1965, revised March 21, 1968, to add Model S2R. Engine installed per STC SA2969WE.

## Required Equipment

The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. In addition, the following equipment is required:

- (1) 24 volt electrical system, Rockwell Dwg. No. 90326.

## Weight and Balance

See NOTE 1 for weight and balance information

## Placards

Remove the following placards previously installed:

- (1) "AVOID CONTINUOUS GROUND OPERATION BETWEEN 1,280 AND 1,900 RPM."
- (2) If alternator was installed:  
"DO NOT TURN OFF ALTERNATOR IN FLIGHT EXCEPT IN CASE OF EMERGENCY"  
"75 AMP MAX." (on left instrument panel)  
"C/B - LAT." (on left instrument panel)
- (3) At fuel filler caps:  
"87 OCTANE"

Add the following placards:

- (1) Adjacent to manifold pressure gage:

	<u>H.P.</u>	<u>R.P.M.</u>	<u>M.P.(In. Hg.)</u>	<u>ALT.</u>
Takeoff (1 min.)	800	2,600	44.0	S.L.
Takeoff (1 min.)	800	2,600	42.5	3,500
Max. Continuous	700	2,400	39.5	S.L.
Max. Continuous	700	2,400	38.0	5,000

Straight line variation between points given.

"100/130 MINIMUM GRADE AVIATION GASOLINE"

- (2) At auxiliary fuel pump/circuit breaker:  
"AUXILIARY FUEL PUMP ON/OFF"
- (3) At primer switch:  
"PRIMER ON/OFF"
- (4) At generator circuit breaker:  
"CB GEN"
- (5) At fuel filler cap:  
"100/130 MINIMUM GRADE AVIATION GASOLINE"
- (6) At altimeter:  
"ALTITUDE LOSS IN STALL RECOVERY - 300 FEET"

NOTE 6: These aircraft have demonstrated satisfactory operation in the Restricted Category under the following conditions:

- (a) Model S2R (with P & W R-1340 Engine) at 6,900 lbs., Standard Day, 400 ft. Altitude, C.G. Limits of 25.0 to 30.0 inches, Stall Speed 78 mph CAS, Maximum Speed 126 mph CAS.
- (b) Model S2R (with Wright R-1300-1B Engine at 7,800 lbs., Standard Day, 1,700 ft. Altitude, C.G. Limits of 24.0 to 28.0 inches, Stall Speed 83 mph CAS, Maximum Speed 126 mph CAS.
- (c) Model S2R-T34 at 8,500 lbs., 2,500 Altitude, Outside Air Temperature 45 °F, C.G. Limits of 30.0 inches, Stall Speed 78 mph CAS with 15° Flaps, Maximum Speed 126 mph CAS.
- (d) Model S2R-T15 at 8,000 lbs., 3,000 ft. Altitude, Outside Air Temperature 65°F, C.G. Limits of 30.0 inches, Stall Speed 76 mph CAS, with 15° Flaps, Maximum Speed 126 mph CAS.
- (e) Deleted.

While items (a) through (d) have been satisfactorily demonstrated, all parts of CAR 3 have not necessarily been complied with for restricted category operations at the increased weights. Also additional operating instructions may need to be established for individual restricted operation approvals under 14 CFR 21.25.

NOTE 7: The following models and serial numbers were produced by the Ayres Corporation (originally Rockwell) at its Albany, Georgia, facility (later serial numbers not listed below were manufactured after July 2003 by Thrush Aircraft, Inc.):

1. Model S2R (600 HP), S/N 1526 through 3002
2. Model S2R (800 HP), S/N 5000 through 5100
3. Model S2R-T34, S/N 6000 through 6049, T34-001 through T34-272 with or without DC suffix, S/N T41-090DC through T41-225DC, and T42-209DC
4. Model S2R-T15, S/N T15-001 through T15-044
5. Model S2R-R3S, S/N R3S-001 through R3S-011, R3S-009DC through R3S-010DC
6. Model S2R-T11, S/N T11-001 through T11-005
7. Model S2R-R1340, S/N R1340-001DC through R1340-010DC, S/N R1340-011 through R1340-035
8. Model S2R-R1820, S/N R1820-001DC through R1820-032DC, R1820-033 through R1820-036
9. Model S2R-T65, S/N T65-001DC through T65-018DC (T65-010DC converted to S2RHG-T65)
10. Model S2RHG-T65, S/N T65-002DC through T65-010DC
11. Model S2R-T45, S/N T45-001DC through T45-014DC, T45-008 through T45-015
12. Model S2R-G6, S/N G6-101 through G6-155, G6-116DC through G6-151DC
13. Model S2R-G10, S/N G10-101 through G10-168, G10-106DC through G10-165DC
14. Model S2R-G5, S/N G5-101 through G5-105
15. Model S2R-G1, S/N G1-101 through G1-115
16. Model S2RHG-T34, S/N T34HG-101DC through T34HG-102DC
17. Model S2R-T660, S/N T660-101 through T660-108

NOTE 8: For Models S2R-R3S, S2R-T34, S2R-T15, S2R-T11, S2R-G6, S2R-G10, S2R-G5 and S2R-T660 with the serial number suffixed with "DC" (Dual Cockpit), the following data apply. All other data listed for these models remain unchanged.

Model S2R-R3S Dual Cockpit, 2PCLM

C.G. Range	Forward limit at 6,000 lbs. is 22.5 inches Aft of Datum. Aft limits at 6,000 lbs. are 27.5 inches Aft of Datum without P/N 19661-1 (elevator down spring) installed. 30.0 inches Aft of Datum with P/N 19661-1 (elevator down spring) installed.
Number of Seats	1 (+89), 1 (+127 Forward Facing) or (+111 Aft Facing)
Maximum Cargo Load	See weight and balance data. Maximum passenger/cargo compartment, 200 lbs. (+120). Maximum hopper load, 3336 lbs. (+29.9).
Control Surface Movements	Flaps Down $15^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	R3S-009DC and subsequent.
Required Equipment	This equipment must include Airplane Flight Manual and Supplement for Restricted Category Operation, dated March 28, 1980, or later approved versions.

Model S2R-T34 Dual Cockpit, 2PCLM

C.G. Range	(+22.5) to (+27.5) without P/N 19661-1 (elevator down spring) installed. (+22.5) to (+30.0) with P/N 19661-1 (elevator down spring) installed.
Number of Seats	1 (+89), 1 (+127 Forward Facing) or (+111 Aft Facing)
Maximum Cargo Load	Passenger/Cargo compartment, 200 lbs. (+120). Maximum hopper load, 3,336 lbs. (+29.9). (See NOTE 10 for increased load limit.)
Control Surface Movements	Elevator Tab $8^{\circ} \pm 1^{\circ}$ up; $22^{\circ} \pm 1^{\circ}$ down
Serial Numbers Eligible	T34-033DC and subsequent.
Required Equipment	This equipment must include Airplane Flight Manual and Supplement for Restricted Category Operation, dated August 25, 1980, or later approved versions.

Model S2R-T15 Dual Cockpit, 2PCLM

Same as S2R-T34 Dual Cockpit except

Serial Numbers Eligible	T15-010DC and subsequent
Required Equipment	This equipment must include Airplane Flight Manual and Supplement for Restricted Category Operation, dated October 7, 1980, or later approved versions.

Model S2R-T11 Dual Cockpit, 2PCLM

Same as S2R-T34 Dual Cockpit

Serial Numbers Eligible	T11-004DC and subsequent
-------------------------	--------------------------

Model S2R-G6 Dual Cockpit, 2PCLM

Number of Seats	1 (+89), 1 (+127)
Maximum Cargo Load	See weight and balance data. Passenger/cargo compartment, 200 lbs. (+120.0)
Serial Numbers Eligible	G6-101DC and subsequent.

Required Equipment This equipment must include Airplane Flight Manual dated February 24, 1994, or later approved versions.

Model S2R-G10 Dual Cockpit, 2PCLM

Number of Seats 1 (+89), 1 (+127)

Maximum Cargo Load See weight and balance data.  
Passenger/cargo compartment, 200 lbs. (+120.0)

Serial Numbers Eligible G10-106DC and subsequent.

Required Equipment This equipment must include Airplane Flight Manual dated August 19, 1994, or later approved versions.

Model S2R-G5 Dual Cockpit, 2PCLM

Number of seats 1 (+89), 1 (+127)

Maximum Cargo Load See weight and balance data.  
Passenger/cargo compartment, 200 lbs. (+120.0)

Serial Numbers Eligible G5-105DC and subsequent.

Required Equipment This equipment must include Airplane Flight Manual dated August 19, 1994, or later approved versions.

Model S2R-T660 Dual Cockpit, 2PCLM

Engine Pratt & Whitney Canada PT6A-65AG, -65AR, -65B (-65AR must have automatic power reserve feature disabled)  
Optional Engine: Pratt & Whitney Canada PT6A-67AG

Engine Limits

	<u>PT6A-65AG/-65AR/-65B:</u>					
	<u>Takeoff</u>	<u>Max. Cont.</u>	<u>Start</u>	<u>Transient*</u>		<u>Idle</u>
				<u>Accel.</u>	<u>Reverse</u>	
SHP	1,300 <sup>†</sup>	1,220 <sup>†</sup>			900	
Torque (PSI)**	48.0	45.1		61*		
ITT (°C)	810	810	1,000*	850*	760	750 <sup>††</sup>
Ng (%)	104	104		104		56
Np (RPM)	1,700	1,700		1,870	1,650	
Oil Press (PSIG)	90 to 135	90 to 135	0 to 200	40 to 200	90 to 135	60 min.
Oil Temp (°C)	0 to 110 <sup>†††</sup>	0 to 110 <sup>†††</sup>	-40 min.	-40 to 110	0 to 110 <sup>†</sup>	-40 to 110

\*Transient engine limits are 5 seconds for starting and 20 seconds for acceleration.

\*\*The Torque pressure limits listed are for NP=1,700 RPM only.

<sup>†</sup> For PT6A-65B 1,100 SHP Take off & Max. Continuous, Torque 43.34 psi Takeoff & max continuous, ITT 820°C Take off; 0 to 99 (°C) Oil Temp Maximum Reverse. For PT6A-65AR ITT 810°C Take off, 810°C Max continuous, 10 to 105 (°C) Oil Temp Maximum Reverse.

<sup>††</sup> For PT6A-65B 700°C, For PT6A-65AR 715°C.

<sup>†††</sup> For PT6-65AR 10 to 110.

PT6A-67AG: Engine Limits same as shown in Section XVII

Propeller and Propeller Limits	For PT6A-65AG/-65AR/-65B: Hartzell HC-B5MP-3F propeller, constant speed, feathering and reversing; Hub Model HC-B5MP-3F with Blade Model M11276NS Diameter 115.2 inches maximum, 114.7 inches minimum. Pitch (42 in. Sta.) 13.9° low, 83.1° feather, -10.0° reverse		
	For PT6A-67AG: Same as shown in Section XVII		
Number of seats	1 (+89), 1 (+127)		
Control Surface Movements	Rudder	Left 20° ± 1°	Right 20° ± 1°
Serial Numbers Eligible	T660-114DC and subsequent		
Required Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification. This equipment must include Thrush Aircraft, Inc. Airplane Flight Manual Dual Cockpit dated August 12, 2005, or later FAA approved revision, is required.		
Agricultural Dispersal Equipment	Standard Spray System, Ayres Dwg. No. 95340 Transland Hydraulic Fire Door Installation, Ayres Dwg. No. 95385		

NOTE 9. The following table summarizes increased fuel capacity limits for the models and serial numbers listed:

<u>Configuration*</u>	<u>A.</u>	<u>B.</u>
	<u>S/N</u>	<u>S/N</u>
<u>Model</u>		
S2R	2564R-4999R	2577R-4999R
S2R-R3S	N/A	N/A
S2R-R1340	R1340-006 to R1340-035	R1340-010 to R1340-035
S2R-T34	T34-034 and up	T34-080 and up
S2R-T15	T15-010 and up	T15-021 and up
S2R-T11	T11-004 and up	T11-004 and up
S2R-R1820	R1820-001 and up	R1820-032 and up

\*Configuration:

- A. Eligible for optional installation  
-190 gallons usable, one 96 gallon tank in each wing, tanks interconnected;  
standard on S2R-R1820.
- B. Eligible for optional installation  
-228 gallons usable, one 115 gallon tank in each wing, tanks interconnected.

NOTE 10. The following table summarizes models and serial numbers eligible for increased hopper load limits - Restricted Category operation only. The certificated maximum take-off gross weight of 6,000 pounds applies to these models and serial numbers.

Hopper Load Limit 4,000 lbs.

<u>Model</u>	<u>S/N</u>
S2R-T34	T34-082 and up
S2R-T15	T15-020 and up
S2R-T11	T11-006 and up
S2R-R1820	R1820-033 and up

NOTE 11. Use of the Ayres P/N 20500 and 20511 (optional) wing tip extension is limited to Restricted Category operation only.

NOTE 12. The Pratt & Whitney (United Aircraft of) Canada PT6A-34 engine is approved as an optional engine on Model S2R-T34, S/N T34-084 and up, and Model S2RHG-T34, S/N T34HG-101 and up, when installed in accordance with Ayres Dwg. 19870 with the following additions:

- A. Bleed Air Case Assy. P/N 3029769; Ref. Pratt & Whitney Service Bulletins 1278 & 1279.
- B. P-3 Air Filter Installation, Ref. Pratt & Whitney Service Bulletins 1253 & 3106.
- C. Replacement of Compressor Delivery Heated Air Tube by a non-metallic hose, P/N 3026687; Ref. Pratt & Whitney Service Bulletin 1315.

Due to the anticipated operating environment, servicing and overhaul interval shall be in accordance with Pratt & Whitney's recommendations for the PT6A-34AG engine.

NOTE 13. The following table summarizes models and serial numbers eligible for Ayres P/N 40220 metal tail as an optional installation if a fabric tail was originally installed by the factory<sup>\*\*</sup>:

<u>Model**</u>	<u>S/N</u>
S2R	1416R and up
S2R-R1340	R1340-001 to R1340-035*
S2R-R1820	R1820-001 and up*
S2R-T34	6001-6049, T34-001 and up*, T41-001 and up*, T15-001 and up*, T27-001 and up*
S2R-T15	T11-001 and up*
S2R-T11	T65-001DC and up
S2R-T65	T45-001DC and up*
S2R-T45	

\*S/N with or without DC suffix

\*\*All other models or serial numbers on this data sheet are eligible for the metal tail only (except the 600 S2D).

Control surface movements (rudder only) for metal tail installations on models listed in Note 13 are the same as for the Model S2R-R1820.

NOTE 14. Models S2R-T34 and S2RHG-T34, Optional Engine Installation  
(Only sections different from Sections III and XVI are shown.)

Engine Pratt & Whitney (United Aircraft of) Canada PT6A-41AG, PT6A-41, or PT6A-42

Due to the anticipated operating environment, servicing and overhaul interval shall be in accordance with Pratt & Whitney's recommendations for the PT6A-41AG engine for the PT6A-41, PT6A-41AG, and PT6A-42 engines.

Fuel PT6A-41AG same requirements as Section III.  
PT6A-41 and PT6A-42 same requirements as Section III except use of Automotive Diesel Number 1D and 2D is prohibited.

Engine Limits for PT6A-41AG, PT6A-41, and PT6A-42:

	<u>Takeoff and Max. Cont.</u>	<u>Transient Start/Accel.</u>	<u>Reverse</u>	<u>Idle</u>
SHP	750		750	
Torque (PSI)	64.5	68.4 Trans	64.5	
ITT (°C)	750	850	750	660
Ng (%)	101.5	102.6	101.5	
Np (RPM)	2,000	2,200	2,000	
Oil Press (PSIG)	105 to 135		105 to 135	60 min.
Oil Temp (°C)	10 to 99	0 to 99.	0 to 99	-40 to 99

Number of Seats	1 (+89) for S/N's without DC suffix; 1 (+89) and 1 (+127) for S/N's with DC suffix.
Serial Numbers Eligible	6000-6049, T41-089 and up, T42-089 and up, T41HG-101 and up, T42HG-101 and up
Placards	Located adjacent to the torque meter: "Maximum Torque is 64.5 PSI at 2,000 RPM"

NOTE 15. For the Models listed in this note with a serial number without a "DC" suffix, the following Single Cockpit configuration data apply; all other data listed for these models remain unchanged:

Model S2R-R1340, Optional Single Cockpit Configuration, 1 PCLM  
(Only sections different from VII are shown.)

Number of Seats	1 (+89)
Maximum Cargo Load	Maximum baggage compartment, 60 lbs. (+112)
Serial Numbers Eligible	R1340-011 and subsequent
Required Equipment	This equipment must include Ayres Airplane Flight Manual dated April 25, 1995, for S/N R1340-001 to R1340-035, or Thrush Airplane Flight Manual dated November 29, 2007 for S/N R1340-036 and up, or later approved versions.

Model S2R-R1820 Optional Single Cockpit Configuration, 1 PCLM  
(Only sections different from VIII are shown.)

C.G. Range	(+23.0) to (+27.5) without P/N 19661 Elevator Down Spring Assy. Installed (+23.0) to (+30.0) with P/N 19661 Elevator Down Spring Assy. Installed
Number of Seats	1 (+89)
Maximum Cargo Load	Maximum baggage compartment, 60 lbs. (+112)
Oil Capacity	18 gallons at Station (-12)
Serial Numbers Eligible	R1820-033 and subsequent
Required Equipment	This equipment must include Airplane Flight Manual dated April 28, 1989, or later approved versions.

Model S2R-T45 Optional Single Cockpit Configuration, 1 PCLM  
(Only sections different from XI are shown)

Number of Seats	1 (+89)
Maximum Cargo Load	See weight and balance data. Maximum baggage compartment 60 lbs. (+112).
Control Surface Movements	Rudder                      Left $19^{\circ} \pm 1^{\circ}$ Right $19^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	T45-001 and subsequent.
Required Equipment	This equipment must include Airplane Flight Manual dated April 25, 1995, or later approved versions.

Model S2RHG-T34, Optional Single Cockpit Configuration, 1 PCLM  
(Only sections different from XVI are shown.)

Number of Seats	1 (+89)
Maximum Cargo Load	Maximum baggage compartment, 60 lbs. (+112)
Serial Numbers Eligible	T34HG-103 and subsequent

NOTE 16. Deleted. (Combined with Note 15)

NOTE 17. Any one of the following agricultural dispersal systems may be installed on Models S2R-T34, S2R-T15, S2R-R3S, S2R-T11, S2R-R1340, S2R-R1820, S2R-T65, S2R-T45, S2R-G6, S2R-G10, S2R-G5, S2R-G1 and S2RHG-T34 all serial numbers, and S2RHG-T65, S/N T65-002DC thru T65-018DC:

- (a) Micronair Spray System, Aero Commander Dwg. No. 80870.
- (b) Boommaster Installation, Aero Commander Dwg. No. 80931.
- (c) Standard Spray System, Rockwell Dwg. No. 81071.
- (d) Spreader and Spreader Quick-Disconnect Installation, Rockwell Dwg. No. 80975.
- (e) Spray System Installation, Rockwell Dwg. No. 80854.
- (f) Fire Bomber System Installation, Rockwell Dwg. No. 81069.
- (g) Spreader and Calibration Installation, Aero Commander Dwg. No. 80674.

NOTE 18. For Models S2R-T34, S2R-T15 and S2R-T11 equipped with optional Air Inlet Barrier Filter, P/N 21402, refer to the Airplane Flight Manual Supplement approved February 4, 1991, or later approved revision, for limitations and procedures.

NOTE 19. External pitot type engine air inlet, P/N's 21900-1 and 21900-21, or screened fairing panel, P/N 21922, are approved optional equipment on the following models:

<u>Model</u>	<u>S/N</u>
S2R-T34	T34-150 and up
S2R-T15	T15-028 and up
S2R-T11	T11-006 and up

NOTE 20. Model S2R, S/N 2584R and subsequent (Diet Thrush)

The following major components have been reduced in weight and structural strength:

<u>ASSEMBLY</u>	<u>PART NUMBER</u>
Wings	20209-600 L/R
Fuselage	10601-600
Horizontal Stabilizer	40087-100

These components are identified at the time of manufacture with the part numbers listed above. This weight and strength reduction effectively reverts these assemblies to their 1977 capabilities. Ayres Corporation will use these components only on Ayres models with a hopper capacity of 400 gallons or less and with engines that are rated at no more than 680 SHP.

NOTE 21. Deleted. (Combined with Note 15.)

NOTE 22. Lower Spar Caps, P/N's 22507T001 and 22507T002, are life limited and must be replaced at 29,000 hours time in service. These P/N's are installed as original equipment on the following serial numbers:

T34/41-271 and up	G6-156 and up
T15/27-041 and up	G10-166 and up
T11-006 and up	G5-106 and up
T65-019 and up	G1-116 and up
T65HG-011 and up	T34HG-103 and up
T45-016 and up	

NOTE 23. The Maximum Weight of the S2R-T660 may be increased from 12,500 pounds to 14,150 pounds if operated in accordance with the limitations shown in the Airplane Flight Manual including airspeed indicator marking changes (for example, Vne is decreased from 219 mph to 160 mph). Applicable Airplane Flight Manuals for operations above 12,500 pounds are Revision 3 of the Ayres S2R-T660 Airplane Flight Manual, or later FAA approved revision, or Thrush Aircraft Inc. S2R-T660 Airplane Flight Manual for serial number T660-109 and up, dated March 5, 2004, or later FAA approved revision, or Thrush Aircraft Inc. S2R-T660 Airplane Flight Manual Dual Cockpit for serial number T660-114DC and up, dated August 12, 2005. For operations in the United States above 12,500 pounds, an exemption to the pilot type rating requirements of 14 CFR Part 61.31(a)(1) obtained by the manufacturer is required. Equivalent Level of Safety Finding No. ACE-04-05 to 14 CFR Part 23.473(b) is applicable regarding the landing weight being less than the takeoff weight.

If operated at weights between 12,500 pounds and 14,150 lbs. the structural limitations shown in Section XVII of this type certificate data sheet are decreased and the following parts must be replaced at the time in service indicated below:

<u>Part Name</u>	<u>Part Number</u>	<u>Life Limit</u>
Rear Spar Doubler, Lower	95627-3	11,000
Rear Spar, Inboard, L&R	95623-1/-2	11,000
Aft Main Spar Lug, L&R	95605-1/-2	11,000
Forward Main Spar Lug, L&R	95606-1/-2	11,000
Spar Cap Assy, L&R	95603-1/-2	13,680
Steel Doubler Plate	95614-1	19,700

NOTE 24. Propeller Pitch Limits  
 Hartzell HC-B3TN-3C and -3D with T10282N+4 blades at 30 inch station:  
 Reverse  $-8.0^{\circ} \pm 0.5^{\circ}$ , Low  $18.0^{\circ} \pm 0.1^{\circ}$ , Feather  $87.0^{\circ} \pm 1.0^{\circ}$   
 Hartzell HC-B3TN-3C with T10282N blades at 30 inch station:  
 Reverse  $-11.0^{\circ} \pm 1.0^{\circ}$ , Low  $16.0^{\circ} \pm 0.1^{\circ}$ , Feather  $89.0^{\circ} \pm 1.0^{\circ}$   
 Hartzell HC-B5MP-3C with M10876ANS or M10876AS blades at 42 inch station:  
 Reverse  $-11.0^{\circ} \pm 0.5^{\circ}$ , Low  $11.5^{\circ} \pm 0.1^{\circ}$ , Feather  $79.0^{\circ} \pm 0.5^{\circ}$   
 Hartzell HC-B4TN-5NL with LT10890N blades at 42 inch station:  
 Reverse  $-6.0^{\circ} \pm 0.5^{\circ}$ , Low  $-1.4^{\circ} \pm 0.1^{\circ}$ , Feather  $81.2^{\circ} \pm 0.5^{\circ}$   
 Hartzell HC-B4TN-5M with T10282N+4 blades at 30 inch station:  
 Reverse  $-6.0^{\circ} \pm 0.5^{\circ}$ , Low  $2.0^{\circ} \pm 0.1^{\circ}$ , Feather  $89.0^{\circ} \pm 0.5^{\circ}$   
 McCauley 4HFR34C653 with -(X/X)-L106FA-0 blades at 30 inch station:  
 Reverse  $-4.0^{\circ} \pm 0.2^{\circ}$ , Low  $12.0^{\circ} \pm 0.5^{\circ}$ , Feather  $87.4^{\circ} \pm 0.2^{\circ}$   
 McCauley X4HFR34C662 with /XL108FA-0 blades at 30 inch station:  
 Reverse  $-4.0^{\circ} \pm 0.2^{\circ}$ , Low  $12.0^{\circ} \pm 0.5^{\circ}$ , Feather  $87.4^{\circ} \pm 0.2^{\circ}$

NOTE 25. Model S2RHG-T65, Optional Engine Installation and single cockpit configuration, 1 PCLM  
 (Only sections different from Section X are shown.)

Engine	Pratt & Whitney Canada PT6A-45A, -45B, -45R
	Pratt & Whitney Canada PT6A-60AG
	Honeywell (Garrett) TPE331-10

## Engine Limits

PT6A-45A, -45B, -45R(shown):

	<u>Takeoff</u>	<u>Max. Continuous</u>	<u>Transient Start/Accel.</u>	<u>Idle</u>
SHP	1,050	1,020		
Torque (PSI) (2 sec.)	38.8	37.7	61.0 Trans	
ITT (°C)	800	800	1,000 Start (5 sec.)	750
Ng (%)	104	104	104	56
Np (RPM)	1,700	1,700	1,870 Trans (5 sec.)	
Oil Press (PSIG)	90 to 135	90 to 135	40 to 200	60 min.
Oil Temp (°C)	0 to 110	0 to 110	99 to 110	-40 to 110

PT6A-60AG:

	<u>Takeoff</u>	<u>Max. Continuous</u>	<u>Transient Start/Accel.</u>	<u>Idle</u>
SHP	1,050	1,020		
Torque (PSI) (2 sec.)	38.8	37.7		
ITT (°C)	820	775	1,000 Start (5 sec.)	750
Ng (%)	104	104	104	58
Np (RPM)	1,700	1,700	1,870 Trans (5 sec.)	
Oil Press (PSIG)	90 to 135	90 to 135	40 to 200	60 min.
Oil Temp (°C)	10 to 110	0 to 110	0 to 110	-40 to 110

TPE331-10:

	<u>Takeoff</u>	<u>Max. Continuous</u>	<u>Transient Start/Accel.</u>	<u>Idle</u>
SHP	900	900		
Torque (PSI)	*	*		
EGT (°C)	**	**	770 Start (5 sec.)	
RPM% †	100	100		72 to 85
Oil Press (PSIG)	70 to 120	70 to 120	40 to 120	
Oil Temp (°C) ‡	55 to 127	55 to 110	-40 to 110	-40 to 110

\*Takeoff torque limit is 2,972 lb/ft at 100%RPM. See current engine logbook for PSI equivalent.

\*\*EGT Limits: 600°C at 45 °C OAT, 540 °C EGT at -15 °C OAT, straight line variation between points.

† Avoid operation between 18% and 28% rpm, except for transient during start and shut down.

‡ Limitation for MIL-L-23699B oil

## Certification Basis

Compliance with FAA Policy Memorandum dated December 1, 1997, Section 23.49, has been shown (61 knot stall speed met with hopper empty), in lieu of previously required Exemption No. 4898.

Equivalent Level of Safety Finding No. ACE-04-05 dated July 26, 2004 for 8,800 pound landing weight.

## C. G. Range

For S/N T65HG-011 and subsequent: Aft limit +29.0 inches aft of datum, at all weights.

For S/N T65HG-015G and subsequent: Aft limit +28.0 inches aft of datum, at all weights.

## Maximum Landing Weight

8,800 lbs. (Aircraft equipped with any engine above and Main Landing Gear p/n 94200 light weight spring gear)  
10,500 lbs. (Only on S/N T65HG-015G and subsequent aircraft which have the

	Honeywell (Garrett) TPE331-10 engine and when equipped with optional Main Landing Gear p/n 94140 heavy weight spring gear. See Airplane Flight Manual for description of gear.)			
Number of Seats	1 (+89)			
Maximum Cargo Load	Maximum baggage compartment, 60 lbs. (+112)			
Control Surface Movements	S/N T65HG-011 and subsequent:	Rudder	Left $22^{\circ} \pm 1^{\circ}$	Right $22^{\circ} \pm 1^{\circ}$
	S/N T65HG-015G and subsequent:	Rudder	Left $19^{\circ} \pm 1^{\circ}$	Right $19^{\circ} \pm 1^{\circ}$
Serial Numbers Eligible	S/N T65HG-011 and subsequent S/N T65HG-015G and subsequent			
Required Equipment	Thrush Aircraft, Inc. S2RHG-T65 Airplane Flight Manual dated July 26, 2004, for S/N T65HG-011 and subsequent; or Thrush Aircraft, Inc. S2RHG-T65 with TPE-331-10 Engine Airplane Flight Manual dated July 22, 2005, for S/N T65HG-015G and subsequent or later FAA approved revision is required.			
Agricultural Dispersal Equipment	Standard Spray System, Thrush Dwg. No. 81071.			

.....END.....