

FEDERAL AVIATION AGENCY

AIEU
BOLKOW
Bolkow Jr.
June 12, 1964

TYPE CERTIFICATE DATA SHEET NO. A1EU

This data sheet which is a part of type certificate No. A1EU prescribes conditions and limitations under which the product for which the type certificate was issued meets the airworthiness requirements of the Civil Air Regulations.

Type Certificate Holder BOLKOW-APPARATEBAU GMBH
Nabern/Teck, Wurttemberg
Federal Republic of Germany

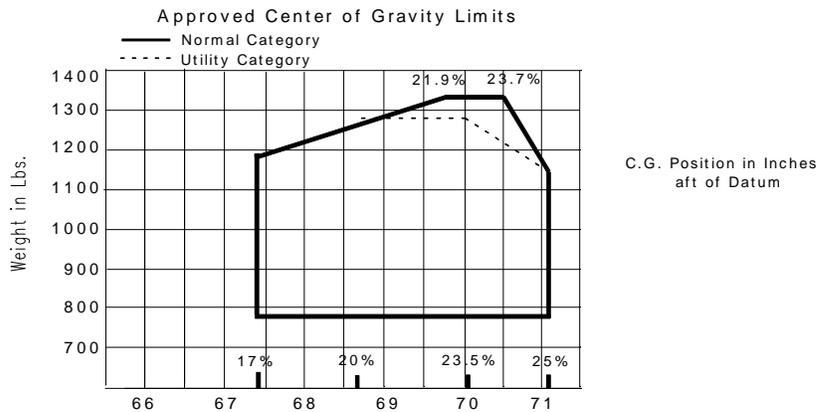
Model Bolkow Junior 2 PCLM (Normal and Utility Category), Approved January 17, 1964

Engine Continental O-200-A
Fuel 80/87 Minimum grade aviation gasoline
Engine limits For all operations, 2750 r.p.m. (100 hp.)

Propeller and McCauley 1A100 MCM 6758
propeller limits Diameter: 67 inches (No cutoff permitted)
Static r.p.m. at maximum permissible throttle setting:
Not over 2400, not under 2300
No additional tolerance permitted.

Airspeed limits (CAS) Vne (Never exceed) 176 mph
Vno (Max. structural cruising) 143 mph
Vp (Maneuvering) 122 mph
Vfe (Flaps extended) 91 mph

Empty weight C.G. range None



Datum 75 inch in front of reference point (reference point: red encircled rivet on each fuselage side wall, located 4 inch forward of hole center of lower wing strut fitting)

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Leveling means	Leveling marks on port fuselage side wall		
Maximum weight	1325 lbs. Normal category, 1270 lbs. Utility category		
No. of seats	2 (+67)		
Maximum baggage	45 lbs. (+88)		
Fuel capacity	21 US gal. (20.5 gal. usable) (+88)		
Oil capacity	5 US qt. (3 qt. usable) (+27)		
Control surface movements	Wing flaps		Down 35°
	Aileron	Up 25°	Down 12°
	Elevator	Up 16°	Down 9°
	Rudder	Right 20°	Left 20°
	Elevator tab	Up 22°	Down 14°

Serial Nos. eligible The Federal Republic of Germany Certificate of Airworthiness for Export endorsed as noted below under "Certification basis" must be submitted for each individual aircraft for which application for certification is made.

Certification basis CAR 10. Type Certificate No. A1EU issued January 17, 1964
Date of Application for Type Certificate November 2, 1962

A US Airworthiness Certificate may be issued on the basis of a Certificate of Airworthiness for Export signed by a representative of the Luftfahrt-Bundesamt, containing the following statement:
"The airplane covered by this certificate has been examined and found to comply with U.S. Civil Air Regulation Part 3, dated May 15, 1958, including amendments 3-1 through 3-7 and conform to T.C. A1EU.

Equipment The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the aircraft for certification. In addition, the following items of equipment are required:

- (a) Pre-stall warning indicator, Safe Flight Instrument Corp. Part No. C-75201 and 1-02-019.
- (b) Shoulder harness
- (c) Approved flight manual dated August, 1963.

NOTE 1. Current weight and balance report including list of equipment included in certificated empty weight, and loading instructions when necessary, must be in each aircraft at the time of original certification.

NOTE 2. The following placard must be displayed on the instrument panel in full view of the pilot:
"This airplane must be operated as a normal or utility category airplane in compliance with the operating limitations stated in the approved Airplane Flight Manual."

NOTE 3. Aircraft must be assembled and inspected in accordance with Luftfahrt Bundesamt approved Bolkow "Assembly Instructions dated August 1963, Reassembly Inspection Report dated September 1963, and Production Flight Test Report dated August 1963".

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