

Supplemental Type Certificate

Number **SR09257RC**

This certificate, issued to

**Global Spar LLC
1000 85th Avenue SE
Olympia, WA 98501**

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 21.25 of the Federal Aviation Regulations.

Original Product—Type Certificate Number: ** See Associated FAA Approved Model List (AML)
Make: **
Model: **

Description of the Type Design Change: Fabrication of Global Spar LLC Vertical Fin Spar Kit, Part No. VTF-030-800-001, in accordance with Global Helicopter Technology, Inc. (GHTI) Master Drawing List No. VTF-392-110, Revision L, dated December 1, 2011, or later FAA approved revision. Installed in accordance with GHTI Manual No. GHT-VTF-099-UH1, Revision E, dated June 5, 2008, or later FAA approved revision. Maintained in accordance with GHTI Instructions for Continued Airworthiness Document No. GHT-392-105, Revision 4, dated October 4, 2002, or later FAA approved revision.

Limitations and Conditions: Approval of this change in type design applies to rotorcraft models specified on associated AML SR09257RC only, that are equipped with standard tail booms with part number 205-030-899-109, fin assemblies installed. This approval should not be extended to any other rotorcraft of this type, in which other previously approved modifications are incorporated, unless it is determined that the relationship between this change and any other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of the helicopter.

(See Continuation Sheet 3 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: April 26, 2000 *Date reissued:* January 27, 2004; October 29, 2010
Date of issuance: November 13, 2000 *Date amended:* 4/22/2011; 5/18/2011; 12/8/11



By direction of the Administrator

(Signature)

Acting Manager, Seattle Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

DepartmentofTransportation_FederalAviationAdministration

Supplemental Type Certificate

(Continuation Sheet)

Number **SR09257RC**

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Issued: November 13, 2000

Reissued: January 27, 2004; October 29, 2010

Amended: 4/22/2011; 5/18/2011; 12/8/11

Limitations and Conditions continued:

This modification is not eligible for aircraft equipped with Bell Helicopter 212 Tail Rotor or other Tail Rotor Retrofit Systems installed (212-011-701, Hub and Blade), or with alternate horsepower engines installed that are not specified on the Type Certificate Data Sheets listed on the associated FAA approved AML.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. No Rotorcraft Flight Manual Supplement was deemed necessary for this modification.

- END -

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.