

Supplemental Type Certificate

IMPORT

Number ST00838NY

This certificate issued to SIDDHIS Aeronautical Design Corporation
4654 Côte-Vertu Ouest
St. Laurent, Quebec
Canada H4S 1C7

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 25 of the Federal Aviation Regulations.

Original Product -- Type Certificate Number : A21EA

Make: Bombardier

Model: CL-600-1A11, CL-600-2A12, CL-600-2B16 (601-3A, 601-3R, & 604 Variants) and CL-600-2B19

Description of Type Design Change:

The installation of the Fin Mounted Radome / Antenna System (Structural Provisions Only) to be done in accordance with SIDDHIS Aeronautical Design Modification Summary MS05272, dated March 26, 2007, Revision 3, or later Transport Canada approved revision.

Limitations and Conditions :

1. Maintenance must be in accordance with SIDDHIS Aeronautical Design Instructions for Continued Airworthiness ICA05272, dated March 19, 2007, Revision 2, or later Transport Canada accepted revision.
2. The FAA has not certified the equipment for which these provisions are intended. You must get additional FAA approval to install the equipment. The FAA must evaluate the installation to ensure it complies with the applicable airworthiness standards.

(See Continuation Sheet 2)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application : March 30, 1998

Date reissued :

Date of issuance : May 18, 1998

Date amended : November 15, 2002, May 12, 2006,
May 14, 2007



By direction of the Administrator

(Signature)

Anthony Socias
Manager
New York Aircraft Certification Office

(Title)

United States of America
Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate
(Continuation Sheet)

Number **ST00838NY**

Date of Amendment: May 14, 2007

Limitations and Conditions (Continued).

3. The total weight of the antenna and installed equipment shall not exceed 40.0 pounds.
4. Additional height clearance of 6.75 inches is required above the original fin tip fairing.
5. For all models, installation shall be visually inspected for external damage each 1,200 hours and for suspected lightning damage after every flight in severe weather conditions.
6. The installer must determine whether this design change is compatible with previously approved modifications.
7. If the holder agrees to permit another person to use the certificate to alter a product, the holder must give the other person written evidence of that permission.

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