



United States of America  
Department of Transportation  
Federal Aviation Administration

Supplemental Type Certificate  
IMPORT

Number: SR03704NY

This certificate issued to: AIR AMBULANCE TECHNOLOGY GmbH  
LAMPRECHTSHAUSENER STRASSE 65  
5282 RANSHOFEN, AUSTRIA

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 27 of the Federal Aviation Regulations.

Original Product – Type Certificate Number: H88EU  
Make: Airbus Helicopters Deutschland GmbH (AHD)  
Model: EC135P1, EC135T1, EC135P2, EC135T2, EC135P2+, EC135T2+, EC135P3, EC135T3

Description of Type Design Change:

- 1. Installation of Air Ambulance Technology GmbH's P/N 135-25-20-5000-600 Series Air Ambulance Equipment (EMS config) or P/N 135-25-20-5000-900 Series VIP Interior (VIP config) in accordance with CMM-EC135-AAT29 (EMS config) Rev. 0, dated September 2014 or later EASA approved revision or CMM-EC135-AAT61 (VIP config) Rev. 0, dated July 2015 or later EASA approved revision.

(See Description of Type Design Change continued on Page 3 of 3)

Limitations and Conditions:

- 1. The aircraft must be operated in accordance with the "Limitations" section of Flight Manual Supplement FMS AAT 18 (EMS config), Rev. 5, dated April 2016 or later EASA accepted revision, or FMS AAT44 (VIP config), Rev. 1 dated April 2016, or later EASA accepted revisions.
- 2. The installer must determine whether this design change is compatible with previously approved modifications.
- 3. If the holder agrees to permit another person to use the certificate to alter the product, the holder must give the other person written evidence of that permission.
- 4. The Incubator installation referenced by CMM-EC135-AAT13 is not a part of this approval.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, and revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: April 05, 2016

Date Reissued:

Date of Issuance: May 25, 2016

Date Amended:

By Direction of the Administrator

Signature \_\_\_\_\_

Title Gaetano Sciortino  
Manager  
New York Aircraft Certification Office

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. This certificate may be transferred or made available to third persons by licensing agreements in accordance with 14 CFR 21.47. Possession of this Supplemental Type Certificate (STC) document by persons other than the STC holder does not constitute rights to the design data nor to alter an aircraft, aircraft engine, or propeller. The STC's supporting documentation (drawings, instructions, specifications, flight manual supplements, etc.) is the property of the STC holder. An STC holder who allows a person to use the STC to alter an aircraft, aircraft engine, or propeller must provide that person with written permission acceptable to the FAA. (Ref. 14 CFR 21.120).



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*Supplemental Type Certificate*

INSTRUCTIONS: The transfer endorsement below may be used to notify the appropriate FAA Aircraft Certification Office of the transfer of this Supplemental Type Certificate. The FAA will reissue the certificate in the name of the transferee and forward it to him.

*Transfer Endorsement*

*Transfer the ownership of Supplemental Type Certificate Number: SR03704NY*

**To** (Name and address of transferee)

**From** (Name and address of grantor)

Extent of Authority (if licensing agreement):

*Date of transfer:*

*Signature of grantor:* \_\_\_\_\_

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# Supplemental Type Certificate

(Continuation Sheet)

Number: SR03704NY

Date of Issuance: May 25, 2016

Description of Type Design Change (continued):

2. Operation must be in accordance with Flight Manual Supplement FMS AAT 18 (EMS config), Rev. 5, dated April 2016 or later EASA accepted revision, or FMS AAT44 (VIP config), Rev. 1 dated April 2016, or later EASA accepted revision, and is required as part of this installation.
3. Instructions for Continued Airworthiness must be in accordance with CMM-EC135-AAT29 (EMS config) Rev. 0, dated September 2014 or later EASA approved revision or CMM-EC135-AAT61 (VIP config) Rev. 0, dated July 2015 or later EASA approved revision, and is required as part of this installation.

Certification Basis:

Based on 14 CFR §21.115 and §21.101, and FAA Order 8110.48, the certification basis for this modification to the Airbus Helicopters Deutschland GmbH (AHD) Model EC135P1, EC135T1, EC135P2, EC135T2, EC135P2+, EC135T2+, EC135P3, EC135T3 rotorcraft is as follows:

- a) The type certification basis for the Airbus Helicopters Deutschland GmbH (AHD) Model EC135P1, EC135T1, EC135P2, EC135T2, EC135P2+, EC135T2+, EC135P3, EC135T3 rotorcraft as shown on TCDS H88EU (Rev. 10) for parts not changed or not affected by the change.
- b) The type certification basis for the Airbus Helicopters Deutschland GmbH (AHD) Model EC135P1, EC135T1, EC135P2, EC135T2, EC135P2+, EC135T2+, EC135P3, EC135T3 rotorcraft as shown on TCDS H88EU (Rev. 10) for parts changed or affected by the change.

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Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. This certificate may be transferred or made available to third persons by licensing agreements in accordance with 14 CFR 21.47. Possession of this Supplemental Type Certificate (STC) document by persons other than the STC holder does not constitute rights to the design data nor to alter an aircraft, aircraft engine, or propeller. The STC's supporting documentation (drawings, instructions, specifications, flight manual supplements, etc.) is the property of the STC holder. An STC holder who allows a person to use the STC to alter an aircraft, aircraft engine, or propeller must provide that person with written permission acceptable to the FAA. (Ref. 14 CFR 21.120).