United States of America

Department of Transportation — Federal Aviation Administration

Supplemental Type Certificate

Number SA4367NM

This certificate, issued to Aircraft Design, Inc.

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 23 *of the* Federal Aviation *Regulations*, effective 2/1/65, with Amendments 23-1 through 23-6.

Original Product — Type Certificate Number: A17WE

Make: Piper (Aerostar)

Model: PA-60-600, -601, -601P, -602P and -700P

Description of Type Design Change: Installation of Cleveland Model 40-128/30-93D (wheel/brake) assembly in accordance with Aircraft Design, Inc. Installation Instructions No. AD 1000, dated October 15, 1988, or later FAA approved revisions.

Limitations and Conditions: Approval of this change in type design applies to the above model aircraft only. This approval should not be extended to aircraft of this model which other previously approved modifications are incorporated unless it is determine that the relationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upo the airworthiness of that aircraft. A copy of this Certificate must be maintained as part of the permanent records for the modified aircraft.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: August 15, 1988

Date of issuance: November 16, 1988

Date reissued:

Date amended:

By direction of the Administrator

D. L. Rogers

(Signature)

Assistant Manager, Seattle
Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA Form 8110-2 (10-68)