



United States of America
Department of Transportation
Federal Aviation Administration

Supplemental Type Certificate
IMPORT

Number: SR02842NY

This certificate issued to: Alpine Aerotech LP
1260 Industrial Road
Kelowna, British Columbia
V1Z 1G5 Canada

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 7/29 of the Civil Air / Federal Aviation Regulations.

Original Product – Type Certificate Number: H1SW
Make: Bell Helicopter Textron, Inc.
Model: 205A, 205A-1, 205B

Description of Type Design Change:

- 1. Installation of Sliding Door Pop Out Windows in accordance with ICE Designs Co. Drawing Package 07002DP-L, Revision 1, dated December 2, 2009, approved by Transport Canada Civil Aviation (TCCA) on April 13, 2010, or later TCCA approved revisions.
- 2. Maintenance must be in accordance with ICE Designs Co. Instructions for Continued Airworthiness, Component Maintenance Manual No. 07002CMM-L, Revision 1, December 12, 2009 accepted by TCCA April 13, 2010, or later TCCA accepted revisions.

Limitations and Conditions:

- 1. The installer must determine whether this design change is compatible with previously approved modifications.
- 2. If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, and revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: April 14, 2010

Date Reissued: August 23, 2013

Date of Issuance: September 9, 2010

Date Amended:

By Direction of the Administrator

Signature _____

Title Anthony E. Gallo
Acting Manager
New York Aircraft Certification Office

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. This certificate may be transferred or made available to third persons by licensing agreements in accordance with 14 CFR 21.47. Possession of this Supplemental Type Certificate (STC) document by persons other than the STC holder does not constitute rights to the design data nor to alter an aircraft, aircraft engine, or propeller. The STC's supporting documentation (drawings, instructions, specifications, flight manual supplements, etc.) is the property of the STC holder. An STC holder who allows a person to use the STC to alter an aircraft, aircraft engine, or propeller must provide that person with written permission acceptable to the FAA. (Ref. 14 CFR 21.120).



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INSTRUCTIONS: The transfer endorsement below may be used to notify the appropriate FAA Aircraft Certification Office of the transfer of this Supplemental Type Certificate. The FAA will reissue the certificate in the name of the transferee and forward it to him.

Transfer Endorsement

Transfer the ownership of Supplemental Type Certificate Number: SR02842NY

To (Name and address of transferee)

From (Name and address of grantor)

Extent of Authority (if licensing agreement):

Date of transfer:

Signature of grantor: _____

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Supplemental Type Certificate

(Continuation Sheet)

Number: SR02842NY

Date Reissued: August 23, 2013

Certification Basis:

CAR 7 dated August 1, 1956, (Rotorcraft Airworthiness Transport Categories) Amendments 7-1 through 7-4
FAR 29.301 Dated February 1, 1965 (Loads) Amendment 29-1
FAR 29.303 Dated February 1, 1965 (Factor of Safety) Amendment 29-1
FAR 29.305 Dated February 1, 1965 (Strength and Deformation) Amendment 29-1
FAR 29.307 Dated October 17, 1968 (Proof of Structure) Amendment 29-4
FAR 29.337 Dated February 1, 1965 (Limit Maneuvering Load Factor) Amendment 29-1
FAR 29.341 Dated February 1, 1965 (Gust Loads) Amendment 29-1
FAR 29.561 Dated February 1, 1965 (Emergency Landing Conditions) Amendment 29-1
FAR 29.601 Dated February 1, 1965 (Design) Amendment 29-1
FAR 29.603 Dated February 1, 1965 (Materials) Amendment 29-1
FAR 29.605 Dated February 1, 1965 (Fabrication Methods) Amendment 29-1
FAR 29.607 Dated October 27, 1968 (Fasteners) Amendment 29-5
FAR 29.609 Dated February 1, 1965 (Protection of Structure) Amendment 29-1
FAR 29.613 Dated February 1, 1965 (Material Strength Properties and Design Values) Amendment 29-1
FAR 29.783 Dated February 1, 1965 (Doors) Amendment 29-1
FAR 29.803 Dated February 25, 1968 (Emergency Evacuation) Amendment 29-3
FAR 29.807 Dated February 25, 1968 (Passenger Emergency Exits) Amendment 29-3
FAR 29.809 Dated February 25, 1968 (Emergency Exit Arrangement) Amendment 29-3
FAR 29.811 Dated February 25, 1968 (Emergency Exit Marking) Amendment 29-3
FAR 29.853 Dated November 26, 1984 (Compartment Interiors) Amendment 29-23
FAR 29.1529 Dated October 17, 1968 (Rotorcraft Maintenance Manual) Amendment 29-4

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