

WE-102

FEDERAL AVIATION AGENCY

FORM APPROVED, BUDGET BUREAU NO. 04-R015

SUPPLEMENTAL TYPE CERTIFICATE

INSTRUCTIONS - SUBMIT IN TRIPLICATE TO LOCAL  
FAA FLIGHT STANDARDS INSPECTOR.

COPY WILL BE RETURNED TO APPLICANT UPON  
ISSUANCE

1. NAME AND ADDRESS OF APPLICANT

Weatherly Aviation Co., Inc.  
2304 San Felipe Rd.  
Hollister, California

2. SUPPLEMENTAL TYPE CERTIFICATE APPLIED FOR:

AIRCRAFT     ENGINE     PROPELLER

ORIGINAL MODEL DESIGNATION

Fairchild M-62A, M62-B, M62-C

NEW MODEL DESIGNATION (If desired)

C63, WM62-D

3. DESCRIPTION OF CHANGE

Modifications made in accordance with drawings shown on FAA Approved, Sealed  
Weatherly Aviation Company Drawing Lists M-62 or C-63.

Limitations of Applicability: The approval of this change in type design applies  
basically to Fairchild Models M62A, M62B and M62C only. This approval should not  
be extended to other specific airplanes of these models on which other previously  
approved modifications are incorporated, unless it is determined that the inter-  
relationship between this change and any of those other previously approved modifi-  
cations will introduce no adverse effect upon the airworthiness of that airplane.

SEE REVERSE SIDE FOR OTHER LIMITATIONS.

Certification Basis: CAR 8 dated October 1950

4a. WILL DATA BE AVAILABLE FOR SALE OR RELEASE TO OTHER PERSONS?

YES     NO

4b. WILL PARTS BE MANUFACTURED FOR SALE (Ref. CAR 1.55)?

YES     NO

5. SIGNATURE AND TITLE OF APPLICANT

January 4, 1960

DATE OF APPLICATION

*John C. Weatherly*

SIGNATURE

President

TITLE

6. To be Completed by FAA

NATURE AND LOCATION OF DATA

Drawings shown on Drawing List M-62 and C63, and associated data, are on file at  
Weatherly Aviation Company, P. O. Box 710, Hollister, California and at FAA  
Western Region, Aircraft Engineering Division.

ORIGINAL TYPE CERTIFICATE NO.

A-724

FAA APPROVAL

SUPPLEMENTAL TYPE CERTIFICATE NO.

SA4-1362

Charles R. Hawke

Original signed by  
Moses L. Lipox

SIGNATURE

DATE OF APPROVAL 1/3/61

Reissued: 2/27/61  
Reissued: 12/20/63  
Reissued: 2/24/66

Chief, Aircraft Engineering Division

TITLE

Forward C. G. +22.16 Ins.

Rearward C.G. +28.0 Ins.

Datum, 3 ins. forward of metal wing leading edge.

Gross Weight, take-off and landing, 3500 lbs. (CAM 8 Appendix "B" Sec.II)

NOTE: Gross weights in excess of 3500 lbs. may be realized when compliance with CAM 8.10-4 (b), (c) has been shown.