

**FAA Model Appendix (MA) AP37.7-MA0830, Amended 10/15/15; STC Number ST00830SE – 737-700 (SEE NOTE 12)**

<b>Kit Descript.</b>	<b>Master Drawing List</b>	<b>Eligible Airplane Serial No.'s</b>	<b>Airplane Maintenance Manual Supplement</b>	<b>Airplane Flight Manual Supplement</b>	<b>Structural Repair Manual Supplement</b>	<b>Weight and Balance Manual Supplement</b>	<b>Certification Maintenance Req't's and Airworthiness Limitations</b>	<b>Maintenance Planning Data Document Supplement</b>	<b>Configuration Deviation List Supplement</b>
Winglet Installation (700-2 Kit) See Notes 1, 4, 5 & 6	<p>1. Modified <b>before 10/10/2007</b>: AP37.7-0609, Rev New, 6/29/2001 or later FAA approved revision</p> <p>2. Modified <b>on or after 10/10/2007</b>: AP37.9-0609, Rev New, 9/7/2007 or later FAA approved revision</p>	737-7007, Rev IR, 8/2/2001 or later APB approved revision	AP37.7-0600, Rev IR, 6/2001 or later APB approved revision	AP37.7-0601, Rev IR, 8/23/2001 or later FAA approved revision	AP37.7-0602, Rev IR, 6/2001 or later FAA approved revision	AP37.7-0603, Rev IR, 6/2001 or later FAA approved revision	<p>1. Modified <b>before 3/17/04</b>: AP37.7-0604, Rev IR, 6/2001 or AP37.1-0704.2, Rev IR, 11/2003 or later FAA approved revisions</p> <p>2. Modified <b>on or after 3/17/04</b>: AP37.1-0704.2, Rev IR, 11/2003 or later FAA approved revision</p>	<p>1. Modified <b>before 3/17/04</b>: AP37.7-0604, Rev IR, 6/2001 or AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision</p> <p>2. Modified <b>on or after 3/17/04</b>: AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision</p>	AP37-CDLS, Rev IR, 12/5/2001 or later FAA approved revision
Winglet Installation (700-3a Kit) (700-3b Kit) (700-3c Kit) See Notes 1, 4, 5 & 6	<p>1. Modified <b>before 10/10/2007</b>: AP37.7-0620, Rev A, 7/26/2004 or later FAA approved revision</p> <p>2. Modified <b>on or after 10/10/2007</b>: AP37.9-0609, Rev New, 9/7/2007 or later FAA approved revision</p>	<p>737-7012, Rev IR, 5/20/2004 or later APB approved revision</p> <p>737-7008, Rev IR, 3/30/2004 or later APB approved revision</p> <p>737-7009, Rev IR, 3/30/2004 or later APB approved revision</p>	AP37.7-0600, Rev E, 5/2004 or later APB approved revision	AP37.7-0601, Rev IR, 8/23/2001 or later FAA approved revision	AP37.7-0602, Rev IR, 6/2001 or later FAA approved revision	AP37.7-0629, Rev IR, 5/2004 or later FAA approved revision	<p>AP37.1-0704.2, Rev B, 5/2004 or later FAA approved revision</p> <p>D626A001-CMR, 7/2004 or later FAA approved revision</p> <p>D626A001-DTR, 7/2004 or later FAA approved revision</p>	AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision	AP37-CDLS, Rev IR, 12/5/2001 or later FAA approved revision

**FAA Model Appendix (MA) AP37.7-MA0830, Amended 10/15/15; STC Number ST00830SE – 737-700 (SEE NOTE 12)**

<b>Kit Descript.</b>	<b>Master Drawing List</b>	<b>Eligible Airplane Serial No.'s</b>	<b>Airplane Maintenance Manual Supplement</b>	<b>Airplane Flight Manual Supplement</b>	<b>Structural Repair Manual Supplement</b>	<b>Weight and Balance Manual Supplement</b>	<b>Certification Maintenance Req't's and Airworthiness Limitations</b>	<b>Maintenance Planning Data Document Supplement</b>	<b>Configuration Deviation List Supplement</b>
Partial Wing Retrofit Installation (700-1 Kit) See Notes 1, 7 & 8	1. Modified <b>before 10/10/2007:</b> AP37.7-0623, Rev New, 5/10/2004 or later FAA approved revision  2. Modified <b>on or after 10/10/2007:</b> AP37.9-0609-1, Rev New, 9/7/2007 or later FAA approved revision	737-7007, Rev IR, 8/2/2001 or later APB approved revision	AP37.7-0600, Rev IR, 6/2001 or later APB approved revision	N/A See Note 7	AP37.7-0602, Rev IR, 6/2001 or later FAA approved revision	AP37.7-0603, Rev IR, 6/2001 or later FAA approved revision	AP37.1-0704.2, Rev A, 2/2004 or later FAA approved revision	AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision	N/A See Note 7
Winglet Installation (737-IGW-2) (BBJ) Kit See Notes 2, 4, 5 & 9	1. Modified <b>before 10/10/2007:</b> AP37.1-0605 Rev A, 8/14/2000 or later FAA approved revision  2. Modified <b>on or after 10/10/2007:</b> AP37.9-0609, Rev New, 9/7/2007 or later FAA approved revision	737-0007, Rev IR, 7/14/2000 or later APB approved revision	AP37.1-0600, Rev IR, 7/2000 or later APB approved revision	AP37.1-0601, Rev IR, 9/1/2000 or later FAA approved revision	AP37.1-0602, Rev A, 8/2000 or later FAA approved revision	AP37.1-0603, Rev IR, 7/2000 or later FAA approved revision	1. Modified <b>before 3/17/04:</b> AP37.1-0604 Rev IR, 7/2000 or AP37.1-0704.2, Rev IR, 11/2003 or later FAA approved revisions  2. Modified <b>on or after 3/17/04:</b> AP37.1-0704.2, Rev IR, 11/2003 or later FAA approved revision	1. Modified <b>before 3/17/04:</b> AP37.1-0604 Rev IR, 7/2000 or AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision  2. Modified <b>on or after 3/17/04:</b> AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision	AP37-CDLS, Rev IR, 12/5/2001 or later FAA approved revision

**FAA Model Appendix (MA) AP37.7-MA0830, Amended 10/15/15; STC Number ST00830SE – 737-700 (SEE NOTE 12)**

<b>Kit Descript.</b>	<b>Master Drawing List</b>	<b>Eligible Airplane Serial No.'s</b>	<b>Airplane Maintenance Manual Supplement</b>	<b>Airplane Flight Manual Supplement</b>	<b>Structural Repair Manual Supplement</b>	<b>Weight and Balance Manual Supplement</b>	<b>Certification Maintenance Req't's and Airworthiness Limitations</b>	<b>Maintenance Planning Data Document Supplement</b>	<b>Configuration Deviation List Supplement</b>
Winglet Installation (700IGW-3) (BBJ-3) Kit and 700ER See Notes 2, 4, 5 & 9	1. Modified <b>before 10/10/2007:</b> AP37.7-0611 Rev A 9/24/2001 or later FAA approved revision  2. Modified <b>on or after 10/10/2007:</b> AP37.9-0609, Rev New, 9/7/2007 or later FAA approved revision	737-0107, Rev IR, 8/6/2001 or later APB approved revision	AP37.7-0618, Rev IR, 9/2001 or later APB approved revision	AP37.1-0616, Rev IR, 10/4/2001 or later FAA approved revision	AP37.7-0615, Rev IR, 9/2001 or later FAA approved revision	AP37.7-0619, Rev IR, 9/2001 or later FAA approved revision	1. Modified <b>before 3/17/04:</b> AP37.1-0604 Rev IR, 7/2000 or AP37.1-0704.2, Rev IR, 11/2003 or later FAA approved revision  2. Modified <b>on or after 3/17/04:</b> AP37.1-0704.2, Rev IR, 11/2003	1. Modified <b>before 3/17/04:</b> AP37.1-0604 Rev IR, 7/2000 or AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision  2. Modified <b>on or after 3/17/04:</b> AP37.1-0704.1 Rev IR, 11/2003 or later APB approved revision	AP37-CDLS, Rev IR, 12/5/2001 or later FAA approved revision
Split Scimitar Winglet Installation (700IGW-5,-6,-7 Kit) See Notes 3, 5, 9, 10 & 11	AP37.9-0609 Rev M 3/20/2015 or later FAA approved revision	737-0105, Rev IR, 12/1/2014 (700IGW-5 Kit), 737-0108, Rev IR, 8/12/2014 (700IGW-6/-7 Kit) or later APB approved revision	AP37.1-0600, Rev J, Dec 2014 (700IGW-5 Kit), AP37.7-0618, Rev E, August 2014 (700IGW-6/7 Kit), or later APB approved revision	AP37.7-0626, Rev IR, 04/15/15, or later FAA approved revision	AP37.1-0602, Rev Feb 2015, (700IGW-5 Kit), AP37.7-0615, Rev Feb 2015 (700IGW-6/7 Kit), or later FAA approved revision	AP37.1-0603, Rev C 01/21/2015 (700IGW-5 Kit), AP37.7-0619, Rev B, Aug 11, 2014 (700IGW-6/7 Kit), or later FAA approved revision	AP37.1-0704.2, Rev R, Feb 2015 or later FAA approved revision	AP37.1-0704.1 Rev U, Jan 2015 or later APB approved revision	AP37-CDLS, Rev 8, 01/28/2015 or later FAA approved revision
Split Scimitar Winglet Installation (700IGW-9 Kit) See Notes 3, 5, 9, 10 & 11	AP37.9-0609 Rev P 8/31/2015 or later FAA approved revision	737-0109, Rev IR, 05/8/2015, or later APB approved revision	AP37.7-0618, Rev G, July 2015, or later APB approved revision	AP37.7-0626, Rev 1, 07/16/15, or later FAA approved revision	AP37.7-0615, Rev June 2015, or later FAA approved revision	AP37.7-0619, Rev C, June 1, 2015, or later FAA approved revision	AP37.1-0704.2, Rev V, August 2015 or later FAA approved revision	AP37.1-0704.1 Rev W, July 2015 or later APB approved revision	AP37-CDLS, Rev 8, 01/28/2015 or later FAA approved revision

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Split Scimitar Winglet Installation (700-4, -5, -5A, -6, -7, -7B, -7C Kit) See Notes 3, 5, 6 & 11	AP37.9-0609 Rev M 3/20/2015 or later FAA approved revision	737-7016, Rev IR, 01/13/15 (700-6,-7 Kit), 737-7017, Rev IR, 1/13/15 (700-4, -5 Kit), 737-7018, Rev IR, 1/13/15 (700-5A Kit) 737-7019 Rev IR, 1/13/15 (700-7C Kit), 737-7021, Rev IR 1/13/15 (700-7B Kit), or later APB approved revision	AP37.7-0600, Rev R, February 2015, or later APB approved revision	AP37.7-0626, Rev IR, 04/15/15, or later FAA approved revision	AP37.7-0602, Rev Feb 2015, or later FAA approved revision	AP37.7-0603, Rev D, 01/26/15 (700-4,-5 Kit), AP37.7-0629, Rev A, 02/16/15 (700-6,-7 Kit), or later FAA approved revision	AP37.1-0704.2, Rev R, Feb 2015 or later FAA approved revision	AP37.1-0704.1 Rev U, Jan 2015 or later APB approved revision	AP37-CDLS, Rev 8, 01/28/2015 or later FAA approved revision
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**Notes:**

1. The 700-1, -2, -3a, -3b, -3c Kit Certification Basis is:

Based on 14 CFR § 21.115 and § 21.101, and the FAA policy for design changes that are identified as significant in FAA Order 8110.48, the certification basis for the design modifications to the Boeing 737-700 Series airplane is as follows:

- a. The type certification basis for areas **not changed** or **not affected** by the change is shown on TCDS A16WE (Section VII. 737-600, -700, -800)
- b. The modification certification basis for areas **changed** or **affected** by the change is shown on TCDS A16WE (Section IX. 737-900): It is based on Part 25 as amended through Amendment 25-91 with some exceptions, voluntary compliance to later amendments, special conditions, ESF's, exemptions as shown in TCDS A16WE (Section IX, 737-900 as applicable as well as those STC unique items as shown below:

**14 CFR Part 25 Regulations same as noted in TCDS A16WE Section IX. 737-900:**

25.1, 25.21, 25.25, 25.29(a), 25.101(,25.103, 25.105, 25.107, 25.109, 25.113-25.115, 25.117, 25.121, 25.123, 25.125, 25.143, 25.147(a)(c)(e), 25.149(a)-(f)(h), 25.161(a)-(d), 25.171, 25.173, 25.175, 25.177(c)(d), 25.181, 25.201, 25.203, 25.207, 25.231(a, 25.235, 25.237(a), 25.251 - 25.255, 25.301 – 25.305, 25.307(a)(d), 25.321- 25.343, 25.345(a)(b)(d), 25.351 –25.361(b), 25.363, 25.365, 25.367-25.391(a)(b)(d)(e), 25.393, 25.409(b), 25.415, 25.427(a)(b)(d), 25.445(a)-25.459, 25.471(a)(b), 25.473, 25.479(a)(c)(d), 25.483-25.491, 25.493(b)(c)(d), 25.495, 25.499, 25.507(a)(b), 25.509(a)(c)(d), 25.511, 25.519, 25.561, 25.571, 25.581, 25.601 – 25.625(a)-(c), 25.629, 25.651(b), 25.655(a), 25.657, 25.671, 25.672, 25.675, 25.677, 25.681,25.685, 25.697(c)(d), 25.699-25.703, 25.723, 25.863, 25.869(a), 25.901(c), 25.903(c), 25.954(c) 25.1011(b), 25.1301, 25.1309, 25.1316(c)(1), 25.1329, 25.1353, 25.1357,25.1385 - 1397, 25.1401(a)(1)(b), 25.1403, 25.1419, 25.1431, 25.1443(c), 25.1501, 25.1505, 25.1519, 25.1529,25.1533, 25.1541, 25.1545, 25.1555, 25.1581(a), 25.1583, 25.1585, 25.1587, C25.1, G25.1, H25.1 - H25.4

**FAA Model Appendix (MA) AP37.7-MA0830, Amended 10/15/15; STC Number ST00830SE – 737-700 (SEE NOTE 12)**

**14 CFR Part 25 Regulations at a later amendment unique to STC:**

25.111(a)-(d) at 25-98  
25.119 at 25-98  
25.145 at 25-98  
25.233 at 25-98  
25.349 at 25-98  
25.481(a)(c) at 25-98

**Equivalent Safety Findings unique to STC:**

Number SA5884SE-T-A-1 for 25.303, 25.305, 25.307, 25.331, 25.333, 25.337, 25.373; Wing aerodynamic loads and applied factor of safety for maneuvering conditions with the speedbrake wing load alleviation system failed. (700-2, -3a, -3b, -3c kits only)  
Number TD5046SE-T-SE-1 for 21.16, 25.1389(b)(1), 25.1389(b)(2), 25.1391, 25.1393; Forward Position Light System Minimum Intensity. (700-2, -3a, -3b, -3c kits only)  
Number TD5046SE-T-SE-2 for 25.1389(b)(3), 25.1395; Position Light Overlapping Intensities. (700-2, -3a, -3b, -3c kits only)  
Number SP11946SE-T-SE-1 for 25.1397(b) for Aviation Green Chromaticity Requirements.  
Number SA5884SE-T-S-1 for 25.1419 Use of Analysis to Demonstrate Safe Flight in Icing Conditions (700-2, -3a, -3b, -3c kits only)

**14 CFR Part 26 Regulations:**

26.11 at 26-0  
26.47 at 26-1

2. The 700IGW-2, -3 Kit Certification Basis is:

Based on 14 CFR § 21.115 and § 21.101, and the FAA policy for design changes that are identified as significant in FAA Order 8110.48, the certification basis for the design modifications to the Boeing 737-700 Series airplane is as follows:

- a. The type certification basis for areas **not changed** or **not affected** by the change is shown on TCDS A16WE (Section VII. 737-600, -700, -800)
- b. The modification certification basis for areas **changed** or **affected** by the change is based on Amendment 25-93 with some exceptions, voluntary compliance to later amendments, special conditions, ESFs and exemptions as shown in TCDS A16WE (Section VII. 737-600, -700, -800) as applicable as well as those STC unique items as shown below::

**14 CFR Part 25 Regulations at the latest amendment 25-0 through 25-93**

25.1, 25.2, 25.3, 25.5, 25.21, 25.25, 25.27, 25.29(a), 25.31, 25.101, 25.103, 25.105, 25.107(c)(3)(e)(4), 25.109, 25.111, 25.113, 25.115, 25.117, 25.119, 25.121, 25.123, 25.125, 25.143, 25.145, 25.147(a)(c)(e), 25.149(a)-(f)(h), 25.161(a)-(d), 25.171, 25.173, 25.175, 25.177(c)(d), 25.181, 25.201, 25.203, 25.207(a)(b)(c)(e)(f), 25.231(a), 25.233, 25.235, 25.237(a), 25.251, 25.253, 25.255, 25.301, 25.303, 25.305, 25.307(a), 25.321, 25.331, 25.333, 25.335, 25.337, 25.341, 25.343, 25.345, 25.349, 25.351, 25.361, 25.363, 25.367, 25.371, 25.373, 25.391(a)(b)(d)(e), 25.409(b), 25.415, 25.427(a)(b)(d), 25.445, 25.459, 25.471, 25.473, 25.477, 25.479, 25.481(a)(c), 25.483, 25.485, 25.487, 25.489, 25.491, 25.493, 25.495, 25.499, 25.503, 25.507(a)(b), 25.509(a)(c)(d), 25.511, 25.519, 25.561, 25.563, 25.571(a)(b)(e), 25.581, 25.601, 25.603, 25.605, 25.607, 25.609, 25.611, 25.613, 25.619, 25.621, 25.623, 25.625, 25.629, 25.671, 25.672, 25.697(d), 25.723, 25.729(a), 25.863(a)(b)(d), 25.901(c), 25.903(c), 25.933(a), 25.954, 25.1011(b), 25.1301, 25.1309(a)(g), 25.1316(c)(1), 25.1329, 25.1353(a)(b), 25.1385, 25.1387, 25.1389(a)(b), 25.1391, 25.1393, 25.1395, 25.1397, 25.1401, 25.1403, 25.1419, 25.1431, 25.1443(c), 25.1519, 25.1529, 25.1581, 25.1583, 25.1585, 25.1587, C25.1, G25.1, H25.1 - H25.4

**Equivalent Safety Findings unique to STC**

Number SA2923SE-T-S-1 for 21.16, 25.1385(c), 25.1387(a)(d)(e); Aft Position Light Dihedral Angle – Obstructed Visibility (700IGW-2 only)  
Number SA2923SE-T-S-4 for 21.16, 25.1389(b)(3), 25.1391, 25.1395; Equivalent Level of Safety Finding for Forward Position Light System (700IGW-2 only)  
Number SA4726SE-T-SE-1 for 21.16, 25.1389(b)(1), 25.1389(b)(2), 25.1391, 25.1393; Forward Position Light System Minimum Intensity (700IGW-3 only)  
Number SA4726SE-T-SE-2 for 25.1389(b)(3), 25.1395; Position Light Overlapping Intensities (700IGW-3 only)  
Number SP11946SE-T-SE-1 for 25.1397(b) for Aviation Green Chromaticity Requirements.  
Number ST2923SE -T-S-2 for 25.1419 Use of Analysis to Demonstrate Safe Flight in Icing Conditions

**14 CFR Part 26 Regulations**

26.11 at 26-0  
26.47 at 26-1

**FAA Model Appendix (MA) AP37.7-MA0830, Amended 10/15/15; STC Number ST00830SE – 737-700 (SEE NOTE 12)**

3. The 700IGW-5, -6, -7, -9 and 737-700-4, -5, -5A, -6, -7, -7B,- 7C Kit Certification Basis is:

Based on 14 CFR § 21.115 and § 21.101, and the FAA policy for design changes that are identified as significant in FAA Order 8110.48, the certification basis for the design modifications to the Boeing 737-700 Series airplane is as follows:

- a. The type certification basis for areas **not changed** or **not affected** by the change is shown on TCDS A16WE (Section VII. 737-600, -700, -800)
- b. The modification certification basis for areas **changed** or **affected** by the change since the reference application date, February 27, 2013 is based upon Part 25 as amended through Amendment 25-136 with some exceptions, voluntary compliance to later amendments, special conditions, ESF's, exemptions as shown in TCDS A16WE (Section X 737-900ER) as applicable as well as those STC unique items as shown below:

**14 CFR Part 25 Regulations at the latest amendment 25-0 through 25-136**

25.1, 25.21(a)-(f)(g)(2), 25.23 - 25.29, 25.101 - 25.107, 25.111, 25.115 - 25.123, 25.125(a)(b)(1)(b)(2)(i)(b)(3)(b)(4)(b)(5)(c)-(g), 25.143 - 25.237, 25.251 - 25.255, 25.301 - 25.305, 25.307(a), 25.321 - 25.351, 25.367 - 25.373, 25.391(d)(e), 25.427(a)(b)(d), 25.445 - 25.471, 25.479(a)(c)(d), 25.481(a)(c), 25.483 - 25.485, 25.489 - 25.491, 25.493(b)(c)(d), 25.495, 25.503, 25.507(a)(b), 25.581, 25.601 - 25.603, 25.605(a), 25.607 - 25.609, 25.611(a) 25.613(a)(b)(c), 25.619(a), 25.621 - 25.629, 25.671 - 25.672, 25.685, 25.697, 25.863, 25.899, 25.901(c), 25.903(c), 25.954, 25.981(d), 25.1011(b), 25.1301, 25.1309(a)-(e), 25.1316, 25.1329(g), 25.1385 - 1395, 25.1401, 25.1419(a)(d), 25.1443(a)(b)(c), 25.1501, 25.1519, 25.1529, 25.1541, 25.1581 - 25.1587, G25.1, H25.1 - H25.3, H25.5 (a)(1)(b)

**14 CFR Part 25 Regulations at the latest amendment 25-0 through 25-136 – For 737-700IGW-5/-5A only**

25.979(c), 25.975(a)

**14 CFR Part 25 Regulations at an intermediate amendment, per requirements documented in SA12511SE-T Issue Paper G-1 Stage 4 Closed Dated 3/6/15**

25.21(g)(1) at 25-72  
25.125(b)(2)(ii)(A)(B)(C) at 25-108  
25.571(a)(1)(2)(b)(c)(d)(e)(2)(3)(4) at 25-86  
25.571(a)(3)(e)(1) at 25-96  
25.841(a)(2) at 25-38  
25.981(a) at 25-11 + additional requirement of added fasteners being installed with cap sealing  
25.1353 (a)(b)(d) at 25-113  
25.1419(e)-(h) at 25-121  
25.1419(c) at 25-23  
C25.1 at 25-0 + additional requirements  
H25.4(a)(1),(a)(2),(b) at 25-123

**Equivalent Safety Findings Unique to STC:**

Number SA5884SE-T-A-1 for 25.303, 25.305, 25.307, 25.331, 25.333, 25.337, 25.373; Wing aerodynamic loads and applied factor of safety for maneuvering conditions with the speedbrake wing load alleviation system failed (700-4/-5/-5A/-6/-7/-7B/-7C only)  
Number SA12511SE-T-G-6-2 for 25.1387(e) at amendment 25-30; rear position light dihedral angle requirements. (700IGW-5,-7, -9 and 700-5,-5A,-7,-7B,-7C only)  
Number SA12511SE-T-G-6-1 for 25.1419(b) at amendment 25-129; flight testing in measured natural icing conditions. (700IGW-5,-7, -9 and 700-5,-5A,-7,-7B,-7C only)  
Number SP11946SE-T-SE-1 for 25.1397(b) for Aviation Green Chromaticity Requirements

**14 CFR Part 26 Regulations**

26.11 at 26-0  
26.47 at 26-1

**14 CFR Part 36 Regulations**

As amended by 36-20 or 36-28. Each Blended Winglet aircraft's FAA approved Airplane Flight Manual (AFM) will indicate if that particular serial number is compliant with Stage 3 or Stage 4 noise levels

4. STC ST00647SE (Flight Dynamics Head-Up Guidance Display Model 2350) has been determined to be compatible with the above. Please refer to Flight Dynamics Master Drawing List 9701-1001, Revision U, dated July 27, 2001, or later FAA approved revisions. Also refer to the appropriate FAA approved Airplane Flight Manual Supplement for this installation.
5. STC ST00845SE (Flight Dynamics Head-Up Guidance Display Model 4000) has been determined to be compatible with this STC. Please refer to Flight Dynamics Master Drawing List 9701-1010, Revision R, dated October 29, 2001, or later FAA approved revisions. Also refer to the appropriate FAA approved Airplane Flight Manual Supplement for this installation.

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6. The speedbrake load alleviation system must be maintained in accordance with the periodic inspection requirements contained in APB documents AP37.7-0604 or AP37.1-0704.1. See the Fault Isolation Manual (FIM) Supplement, AP37.8-0610, for information regarding the speedbrake load alleviation system.
7. For the Partial Wing Retrofit Installation (700-1 Kit), operations and limitations are the same as for the baseline aircraft and Boeing AFM and CDL data apply.
8. When taking an aircraft previously retrofit with APB winglets defined by MDL AP37.7-0609 or AP37.9-0609:700-2 and returning the aircraft back to the standard wingtip configuration defined within this approved configuration by MDL AP37.7-0623 or AP37.9-0609-1:700-1, use only APB Service Bulletin 737-57-008 respectively as the approved instructions.
9. The mass distribution of STC ST00936NY (PATS Auxiliary Fuel System) as defined on or before September 5, 2000, has been determined to be compatible with the above.
10. STC ST01697SE, Cabin Altitude reversion to 6,500 feet in lieu of 8,000 feet at 41,000 feet flight altitude (Lower Cabin Altitude modification) has been determined to be compatible with this STC. Refer to Figure 1 of Boeing Report D926A200, Revision N, dated May 23, 2009, or later Federal Aviation Administration (FAA) approved revision.
11. STC ST00039MC (Rockwell Collins) for installation of Head-Up Guidance Display Model 4000 with EVS (Enhanced Vision System), reference MDL 9701-1018, revision C, dated June 17, 2010 or later FAA approved revision has been determined to be compatible with this STC. Also refer to the appropriate FAA approved Airplane Flight Manual Supplement for this installation.
12. The format of this STC appendix was revised and combined from the Approved Model List (AML) document (AP37.7-AML0830 for the 737-700& AP37.1-AML0830 for the 737-700IGW) format to the Model Appendix (MA) document (AP37.7-MA0830) format for the 737-700/700IGW model at the 4/21/15 STC amendment.

FAA Approved: \_\_\_\_\_

Manager, Seattle Aircraft Certification Office

Issued: October 10, 2007

Amended: 7/14/11; 6/27/12; 4/21/15; 10/15/15

Reissued: N/A