

Supplemental Type Certificate

Number SA708EA

This Certificate issued to Lockheed Martin Aeronautics Company
36 South Cobb Drive
Marietta, Georgia 30063

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 25 of the Federal Aviation Regulations and Part 4b of the Civil Air Regulations.

Original Product Type Certificate Number: 4A22

Make: Lockheed

Model: 188C

Description of Type Design Change: Installation of Opening in Fuselage R/H for Emergency Exit between F. S. 226 and F. S. 253.4 in accordance with Lockheed Aircraft Service Company Report #3607 dated 10/16/68, Revised 12/12/68.

Limitations and Conditions: This modification is approved only for its structural, electrical and mechanical provisions and can be incorporated with STC SA707EA and STC SA709EA. This approval should not be incorporated in any aircraft of this specific model on which other approved modifications are incorporated, unless it is determined that the interrelationship between this change and any of those previously incorporated approved modifications will not introduce any adverse effect upon the airworthiness of the aircraft. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: November 7, 1968

Date reissued: Mar. 20, 1974, Nov. 2, 1999,
Mar. 16, 2001, Mar. 15, 2004, Apr. 26, 2004

Date of issuance: December 17, 1968

Date amended:



By direction of the Administrator

(Signature)

Manager, Technical & Administrative Support
Staff, Los Angeles Aircraft Certification Office

(Title)